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THREE-DIMENSIONAL RAY TRACE

COMPUTER PROGRAM FOR ELECTROMAGNETIC

WAVE PROPAGATION STUDIES





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THREE-DIMENSIONAL RAY TRACE COMPUTER PROGRAM
FOR ELECTROMAGNETIC WAVE PROPAGATION STUDIES,

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ABSTRACT

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A computer program is described for use on the IBM-704/7090 electronic data processing machine or any large computer accepting FORTRAN. The necessary modifications for use on these two computers are simple and self-evident. The computer program permits the computation of detailed ray patterns portraying the spatial distribution of rays emitted from a transmitter whose geographic coordinates with respect to the center of the earth are known. This program deals with the solution of the differential equations, given by Hamiltonian optics, for ray paths in non-isotropic, three-dimensionally nonhomogeneous media whose complex phase refractive index is given by the Appleton-Hartree formula.

This report is to be considered as a first attempt in presenting an account of the current status of the development of this program, which has yielded many useful results. Presented also are sample calculations as well as some results that have been obtained by using this program.

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::TABLE OF CONTENTS

			Page
	ABSTRACT		iii
	LIST OF ILLUSTRA	ATIONS	vii
	LIST OF TABLES		ix
	SECTION		
	ı	INTRODUCTION	1
	II	COMPUTATIONAL PROCEDURE	3
		A. Ray Trace Equations	3
		B. List of Useful Formulae for Ray Tracing	11
		C. Coordinate Transformation	17
		D. Model lonosphere	31
•		E. Model of Earth's Magnetic Field	33
		F. Model of Atmospheric Collision Frequency	35
ì		G. Computational Results	36
	III	COMPUTER PROGRAM FOR THREE- DIMENSIONAL RAY-TRACING	41
		A. Main Program RAY TRACE	45
		B. Function SLANTR	65
		C. Function QATAN	69
•		D. Function ARCOS	73
		E. Subroutine COORD	77
•		F. Subroutine DAUX	82

RM 61TMP-32

SECTION			Page
III	G.	Subroutines INT and INTM	86
	н.	Subroutine RINDEX	112
	ı.	Subroutine ELECTX	129
	J.	Subroutine BIGR	133
	κ.	Subroutine MAGY	137
	L.	Subroutine COLFRZ	141
	M.	Subroutine RCOORD	145
	N.	Subroutine OUTONE	150
	0.	Subroutine OUTPUT	154
	P.	INPUT-OUTPUT	159
ACKNOWLEDG	EMENTS		167
REFERENCES			169

LIST OF ILLUSTRATIONS

Figure	Title	Page
i a	Geometry of Coordinate Transformation	18
16	Geometry of Coordinate Transformation	19
1c	Geometry of Coordinate Transformation	21
1 d	Geometry of Coordinate Transformation	22
2	Geometry for Starting Point and Geomag- netic Coordinate System	28
3	Geometry for Refraction by Spherically Stratified Region	37
4	Radar Propagation Paths Through Spher- ically Ionized Region	43
5	Elevation and Azimuth Errors for Propaga- tion through a Spherical Model - f = 1k Mc	44
6	Block Diagram of Three-Dimensional Ray- Tracing Program	44 a

LIST OF TABLES

Table	Title	Pag
1	Comparison of Total Ray Bending Angle	38
2	Nomenclature describing the V and W Vectors	48
3	Nomenclature Describing the R Vector	113
4	Input Data for a Spherical Ionosphere	160
5	Output Data on Tape 6 for Debugging Purposes	161
6	Output Data From Tane 10	143

SECTION I

What is customarily referred to as reflection in electromagnetic wave propagation, is actually the result of an integrated effect of a phenomenon of refraction (i. e., the gradual changing of the direction of the electromagnetic energy transport vector). Insofar as the phenomenon of refraction is concerned it is well established that the spatial gradient of electron density plays a crucial role in controlling the propagation behavior of an electromagnetic wave. Although this is well known, propagation studies so frequently incorporate a curious mixture of simple refraction phenomena (in the use of Snell's law) and of mirror-like reflection. Hence, in these studies they omit an accounting of these spatial gradients of electron density as well as their variation. As generally known, these oversimplified studies derive from Snell's law the condition for reflection. that being, that at the spatial point of reflection, the electron density must attain the value of 1.24 x 10^4 f² cos² I electron/cc, where f is the propagation frequency in Mc/sec and I is the angle of incidence of the ray upon the first layer of the assumed stratified ionosphere. Such a simplified approach is necessary because of the difficult task that leads to a numerical solution of a propagation problem which incorporates these electron density gradients in its solution.

With increasing use of transmitters in satellites, as well as, for the understanding of the behavior of wave propagation under severly abnormal atmospheric conditions, it becomes important to take a realistic account of the spatial gradient of the electron density in electromagnetic wave propagation studies. Only in this manner will it become possible to usefully utilize the new satellite propagation techniques in studies designed toward the understanding of the atmospheric ionization-deionization phenomena and through this the detailed structure of the ionosphere.

Within the last ten years some effort has been made in constructing analogue computers for the study of ray propagation which accounted for spatial electron density gradients (as for example D. F. Hartree, et al., Manchester, England; M.S. Wong, AFCRL, Bedford, Mass.).

Some of these analogue computers were and still are limited to spatial electron density gradients in a particular direction thus forcing the propagation problem into a two-dimensional plane, or to the study of the behavior of refraction on wave propagation. These constraints are built into the analogue computer and are not easily changed.

One approach that avoids these constraints is the use of a large electronic data processing system where the ordered logical flow controlling any calculation, is easily varied. Combining such a programmed computer with Hamiltonian optics, which give the desired ray tracing equations for a nonhomogeneous, non-isotropic mediu, and the Appleton-Hartree formula for the complex refractive indes, permits in addition to three dimensional ray tracing, the simultaneous study of numerous other variables of the propagation problem. Such an approach to the ray trace propagation problem is presented in its present state of development. A great deal of improvement in some of the routines is possible. As a result, the writer would like to encourage correspondence concerning these matters. In addition, it should be stated, that a computer program for solving the threedimensional ray trace problem has also been written for the Ferranti Mercury Computer at Manchester University by C. B. Haselgrove and J. Haselgrove. 2

SECTION II COMPUTATIONAL PROCEDURE

A. RAY TRACE EQUATIONS

When electromagnetic waves are propagated through a medium whose dielectric constant or index of refraction is a varying function of the path, the waves undergo a change in direction, or refractive bending, and a retardation in the velocity of propagation. The spatial relationship expressing this angular bending of a ray of an electromagnetic wave can be determined by basing the theory of rays and waves on a variational principle (Fermat's) in space. By a ray is meant the path travelled by the transport vector of electromagnetic wave energy between the transmitter and an associated electromagnetic field-intensity point in space. This Hamilton Theory starts from the variational principle $\delta \int \mu ds = 0$, where μ is a medium function or index of refraction, depending on position and direction. From this principle the theory constructs the properties of systems of rays and the waves associated with them (i.e., extremals and transversals, in the language of the calculus of variations). Because of the stationarity in time, the theory may be regarded as a statical one, the rays being fixed curves in space and the waves fixed surfaces. Neither wave-length nor frequency is involved. Likewise the waves form a continuous set of surfaces, not distinguished as crests and troughs. This theory, whether in the form preferred by Hamilton or otherwise, has been the subject of many books under the general title "Geometric Optics".

Thus, applying Hamiltonian optics leads to the general Hamilton's Equations¹ for ray paths of electromagnetic waves in a three-dimensional non-isotropic nonhomogeneous medium. From them Haselgrove² has derived the following set of equations for ray paths in a spherical coordinate system in a format suitable for numerical integration on high speed computers:

$$\frac{d\mathbf{r}}{d\tau} = \frac{1}{\mu^2} \left(\sigma_{\mathbf{r}} - \mu \frac{\partial \mu}{\partial \sigma_{\mathbf{r}}} \right) \tag{1}$$

$$\frac{d\theta}{d\tau} = \frac{1}{r\mu^2} \left(\sigma_{\theta} - \mu \frac{\partial \mu}{\partial \sigma_{\theta}} \right) \tag{2}$$

$$\frac{d\phi}{d\tau} = \frac{1}{\mu^2 r \sin \theta} \left(\sigma_{\phi} - \mu \frac{\partial \mu}{\partial \sigma_{\phi}} \right)$$
 (3)

$$\frac{d\sigma_{r}}{d\tau} = \frac{1}{\mu} \frac{\partial \mu}{\partial r} + \sigma_{\theta} \frac{d\theta}{d\tau} + \sin \theta \sigma_{\phi} \frac{d\phi}{d\tau}$$
 (4)

$$\frac{d\sigma_{\theta}}{d\tau} = \frac{1}{r} \left[\frac{1}{\mu} \frac{\partial \mu}{\partial \theta} - \sigma_{\theta} \frac{dr}{d\tau} + r \cos \theta \sigma_{\phi} \frac{d\phi}{d\tau} \right]$$
 (5)

$$\frac{d\sigma_{\varphi}}{d\tau} = \frac{1}{r \sin \theta} \left[\frac{1}{\mu} \frac{\partial \mu}{\partial \varphi} - \sin \theta \, \phi \, \frac{dr}{d\tau} - r \cos \theta \, \sigma_{\varphi} \, \frac{d\theta}{d\tau} \right] \tag{6}$$

In these equations r, θ , and Φ are the spatial coordinates of a spherical system; μ is the arbitrary index of refraction; $\vec{\sigma}$ is a vector directed normal to the phase fronts of the ray of magnitude μ with σ_r θ , and σ_{ϕ} its respective components in the r, θ , and Φ directions; τ is the time of phase travel along the ray path (i.e., $f\Delta\tau/c$ = the number of wavelengths in the medium along the ray path, where f is the electromagnetic wave frequency and c the velocity of light in vacuum).

It is noteworthy that the partial derivatives of μ appear explicitly in Equations 1 to 6, in accordance with the fact that the gradients of μ play crucial roles in determining the spatial ray paths.

This closed set of first-order partial differential equations which will describe the propagation behavior of an electromagnetic wave

under geometric optics conditions, can be integrated simultaneously by a point-by-point numerical process if expressions can be developed for the necessary derivatives of the phase refractive index μ . The quantity μ and its derivatives are obtained by using the Appleton-Hartree formula 3 as the definition of the complex phase refractive index M.

The derivatives are derived under the conditions of ray optics, that is, that the imaginary part of M² is very much smaller than the real part. As an aid for computer use and comparison with published works of others², ⁴, the Appleton-Hartree formula is written as:

$$M^{2} = (\mu - j\kappa)^{2} = 1 - \frac{2X(1 - X - jZ)}{D}$$
 (7)

$$D = 2(1 - X - jZ)(1 - jZ) - Y^{2} \sin^{2} \psi + S$$
 (8)

$$S = \pm \left[(Y \sin \psi)^4 + 4Y^2 (1 - X - jZ)^2 \cos^2 \psi \right]^{1/2}$$
 (9)

where

M = the complex phase refractive index

X = a scalar quantity representing the normalized electron density

$$\frac{4\pi Ne^2}{m\omega^2} = \frac{\omega p^2}{\omega^2}$$

 $\omega_{\rm p}$ = plasma frequency at the spatial point

 ω = angular wave frequency = $2\pi f$

m, e = mass and charge of an electron

N = electron density at a spatial point

Y = normalized magnitude of the earth's magnetic field vector \vec{Y} =

$$\frac{\overrightarrow{eH}}{mc\omega} = \frac{\overrightarrow{\omega}_c}{\omega}$$

ω = magnitude of the gyromagnetic frequency of the electron about the earth's magnetic field

Z = normalized collision frequency = (v/ω)

v = collision frequency at a spatial point in collisions per second

 ψ = angle defined by the inner product of the vectors $\vec{\sigma}$ and \vec{Y} =

$$\cos^{-1}\left[\frac{\sigma_{r}Y_{r} + \sigma_{\theta}Y_{\theta} + \sigma_{\phi}Y_{\phi}}{(\mu Y)}\right]$$

 $\kappa = \frac{ck}{\omega} = \text{imaginary part of the complex phase refractive index}$

c = velocity of light in vacuum

k = absorption coefficient of the wave per unit length of path (it is proportional to the conductivity of the medium)

It is noted that there are two possible values for the complex index of refraction M corresponding to the plus and minus sign on S which represent two different modes of ionospheric propagation. These are commonly called "ordinary" and 'extraordinary modes for the plus and minus sign respectively. Also, the Appleton-Hartree formula (Equations 7 to 9) is notable in that it presents μ , which actually is a spatial function of six variables, in the form containing purely algebraic operations on factors or terms each of which is a function of at most three variables, that is, either of r, θ , φ or σ_{θ} , σ_{φ} . This reduces the representation of μ to a numerical problem, easily solvable to current computer techniques.

If i represents any one of the spatial spherical coordinates r, θ , and Φ , then the partial derivatives of μ with respect to the components of the wave normal, $\hat{\sigma}$, can be easily shown to be

$$\frac{\partial \mu}{\partial \sigma_{i}} = \frac{\partial \mu}{\partial \Psi} \frac{\partial \Psi}{\partial \sigma_{i}} = \frac{\partial \mu}{\partial \Psi} \left(\frac{\sigma_{i} Y \cos \Psi - \sigma Y_{i}}{\sigma^{2} Y \sin \Psi} \right)$$
 (10)

This useful transformation also enjoys the following property:

When
$$\psi \to 0$$
, $\frac{\partial \mu}{\partial \psi} \to 0$, $\frac{\partial \psi}{\partial \sigma_i} \to \infty$ but $\frac{\partial \mu}{\partial \sigma_i} \to 0$ (11)

To evaluate $\partial \mu / \partial \sigma_i$; the necessary partial derivative is:

$$\frac{\partial \mu}{\partial \Psi} = \operatorname{Re} \frac{\partial M}{\partial \Psi} = \operatorname{Re} \left\{ \frac{(M^2 - 1) (Y^2 \sin \Psi \cos \Psi)}{MD} \left[1 - \frac{1}{S} \left[(Y \sin \Psi)^2 - 2(1 - X - jZ)^2 \right] \right] \right\}$$

$$\therefore \frac{\partial \mu}{\partial \Psi} = \left\{ - Y^2 \sin \Psi \cos \Psi \left[a_0 (a_2 a_5 - b_2 b_5) - b_0 (a_2 b_5 + b_2 a_5) \right] \right\}$$
(12)

where a_0 , b_0 and all following a_j , b_j are defined in the List of Useful Formulae. The partial derivatives of the real part of the phase refractive index with respect to the spatial coordinates (i = r, θ , or Ψ) are similarly obtained by use of the relationship

$$\frac{\partial \mu}{\partial i} = \frac{\partial \mu}{\partial X} \frac{\partial X}{\partial i} + \frac{\partial \mu}{\partial Y} \frac{\partial Y}{\partial i} + \frac{\partial \mu}{\partial Z} \frac{\partial Z}{\partial i} + \frac{\partial \mu}{\partial Y} \frac{\partial Y}{\partial i}$$
(13)

where

$$\frac{\partial \mu}{\partial X} = \text{Re} \frac{\partial M}{\partial X} = \text{Re} \left\{ \frac{1}{\text{MD}} \left[2X - 1 + jZ + (M^2 - 1) \left(1 - jZ + \frac{2Y^2 (1 - X - jZ) \cos^2 \psi}{S} \right) \right] \right\}$$

$$\therefore \frac{\partial \mu}{\partial X} = \left\{ a_0 \left[(2X - 1) - (a_4 a_5 - b_4 b_5) \right] + b_0 \left[Z + a_5 b_4 + b_5 a_4 \right] \right\}$$
(14)

$$\frac{\partial \mu}{\partial Y} = \operatorname{Re} \frac{\partial M}{\partial Y} = \operatorname{Re} \left\{ \frac{(M^2 - 1)}{MD} Y \left[(\sin \psi)^2 - \frac{1}{S} \left[Y^2 \sin^4 \psi + 2(1 - X - jZ)^2 \cos^2 \psi \right] \right] \right\}$$

$$\therefore \frac{\partial \mu}{\partial Y} = Y \left\{ (a_0 a_5 - b_0 b_5) \left[a_6 - (\sin \beta)^2 \right] - b_6 (a_0 b_5 + b_0 a_5) \right\}$$
(15)

$$\frac{\partial \mu}{\partial Z} = \text{Re} \frac{\partial M}{\partial Z} = -\text{Im} \left\{ \frac{1}{\text{MD}} \left[X + (M^2 - 1) \left(2 - X - 2jZ + \frac{2Y^2 (1 - X - jZ)}{S} \cos^2 \psi \right) \right] \right\}$$

$$\frac{\partial \mu}{\partial Z} = \left[b_0 (X - a_5 a_7 + b_5 b_7) - a_0 (b_5 a_7 + a_5 b_7) \right]$$
(16)

and where Re and Im represent, respectively, the real and imaginary part of the complex expression. The partial derivative of the angle Ψ (which is the angle defined by the inner product of the normalized geomagnetic field vector \vec{Y} and the wave normal vector $\vec{\theta}$), with respect to the spatial coordinates r, θ , and Ψ , measures the change in spatial direction of the earth's magnetic field since the calculation is made holding $\vec{\theta}$ constant. The partial derivatives $\partial X/\partial i$, $\partial Y/\partial i$ and $\partial Z/\partial i$ are obtainable from the analytical expressions for the spatial variation of the electron density, geomagnetic field and collision frequency. Examples of these will be considered later.

In addition to Equations 1 to 6, which define the spatial ray path, it is usually desirable to calculate the optical path length s, the time of travel T, as well as, the one way absorption A, of the energy of an electromagnetic pulse. The equation describing the differential optical path is given by

$$\frac{\mathrm{d}s}{\mathrm{d}\tau} = \frac{1}{\mu^2} \left[\mu^2 + \left(\frac{\partial \mu}{\partial \Psi} \right)^2 \right]^{1/2} \tag{17}$$

In determining the time of travel T, a distinction must be made between two electromagnetic wave velocities. The phase velocity, $\mathbf{v}_p = \mathbf{c}/\mu$, is defined as the spatial velocity with which a point of constant phase moves. Group velocity, $\mathbf{v}_g = \mathbf{d}\omega/\mathbf{d}~(\omega/\mathbf{v}_p)$, is the spatial velocity of electromagnetic energy travel; put into other words, it is the velocity of a "Maxwell Demon" who remains at the same point on the envelope of the advancing wave. From these two definitions it can be easily shown that the time (in seconds) of energy pulse travel can be written as:

$$\frac{dT}{d\tau} = \frac{1}{c} \left[1 + \frac{\omega}{\mu} \frac{\partial \mu}{\partial \omega} \right]; \tag{18}$$

where

$$\frac{\partial \mu}{\partial \omega} = \operatorname{Re} \frac{\partial M}{\partial \omega} =$$

$$- \operatorname{Re} \left\{ \frac{1}{MD\omega} \left[X(2X + jZ) + (M^2 - 1) \left[2 - 2jZ - jZX + \frac{2(Y\cos\psi)^2}{S} (1 - X - jZ)(1 + X) \right] \right\}$$

$$\therefore \frac{\partial \mu}{\partial \omega} = - \frac{1}{\omega} \left[a_0 (2X^2 - a_5 a_8 + b_5 b_8) + b_0 (XZ + b_5 a_8 + b_8 a_5) \right]$$
(19)

The one-way absorption, A (in nepers), suffered by the energy of an electromagnetic pulse is determined by

$$\frac{dA}{d\tau} = -\frac{\omega}{c}\frac{\kappa}{\mu}A = -\frac{k}{\mu}A \tag{20}$$

where k (which is proportional to the spatial conductivity) is the absorption of the wave per unit length of path.

The solution of this set of first order partial differential equations will describe the propagation characteristics of an electromagnetic wave in a heterogeneous anisotropic medium.

B. LIST OF USEFUL FORMULAE FOR RAY TRACING

As an aid for the computer solution of these differential equations the following list of formulae are found to be very useful. As before, Re and Im respectively represent the real and imaginary parts of the complex quantity.

$$ReS = S_1 = R_S \cos \theta_S \tag{21}$$

$$ImS = S_2 = R_S \sin \theta_S$$
 (22)

$$R_{S} = \left\{ \left[(Y \sin \psi)^{4} + (2Y \cos \psi)^{2} \left[(1 - X)^{2} - Z^{2} \right] \right]^{2} + \left[(2Y \cos \psi)^{2} \left[2(1 - X)Z \right] \right]^{2} \right\}$$

$$(23)$$

$$\theta_{S} = \frac{1}{2} \tan^{-1} \left\{ \frac{(2Y \cos \psi)^{2} [2(X-1)Z]}{(Y \sin \psi)^{4} + (2Y \cos \psi)^{2} [(1-X)^{2} - Z^{2}]} \right\}$$
(24)

ReD =
$$d_1 = \left\{ 2 \left[(1 - X) - Z^2 \right] - (Y \sin \psi)^2 + S_1 \right\}$$
 (25a)

ImD =
$$d_2 = [S_2 - 2Z(2 - X)]$$
 (25b)

$$ReM = m_1 = \mu = R_m \cos \theta_m \tag{26}$$

$$ImM = m_2 = -\kappa = R_m \sin \theta_m \tag{27}$$

$$R_{m} \left\{ \left(1 - \frac{2x \left[(1 - x)d_{1} - Zd_{2} \right]^{2}}{d_{1}^{2} + d_{2}^{2}} \right)^{2} + \left(\frac{2x \left[Zd_{1} + (1 - x)d_{2} \right]}{d_{1}^{2} + d_{2}^{2}} \right)^{2} \right\}^{1/4}$$
(28)

$$\theta_{m} = \frac{1}{2} \tan^{-1} \left\{ \frac{2X' \left[Zd_{1} + (1 - X)d_{2} \right]}{d_{1}^{2} + d_{2}^{2} - 2X \left[(1 - X)d_{1} - Zd_{2} \right]} \right\}$$
(29)

$$a_{o} = \frac{(m_{1}d_{1} - m_{2}d_{2})}{(m_{1}d_{1} - m_{2}d_{2})^{2} + (m_{1}d_{2} + m_{2}d_{1})^{2}}$$
(30)

$$b_{o} = \frac{(m_{1}d_{2} + m_{2}d_{1})}{(m_{1}d_{1} - m_{2}d_{2})^{2} + (m_{1}d_{2} + m_{2}d_{1})^{2}}$$
(31)

$$a_1 = \left\{ 2 \left[(1 - X)^2 - Z^2 \right] - (Y \sin \psi)^2 \right\}$$
 (32)

$$b_1 = 4Z(1 - X)$$
 (33)

$$a_2 = \left[1 + \frac{(a_1S_1 - b_1S_2)}{S_1^2 + S_2^2}\right]$$
 (34)

$$b_2 = \left[\frac{s_1 b_1 + a_1 s_2}{s_1^2 + s_2^2} \right]$$
 (35)

$$a_{4} = \left\{ 1 + \frac{2(Y\cos\psi)^{2}}{S_{1}^{2} + S_{2}^{2}} \left[S_{1}(1-X) - ZS_{2} \right] \right\}$$

$$b_{4} = \left\{ Z + \frac{2(Y\cos\psi)^{2}}{S_{1}^{2} + S_{2}^{2}} \left[S_{2}(1-X) + ZS_{1} \right] \right\}$$
(36)

$$b_4 = \left\{ Z + \frac{2(Y\cos\psi)^2}{S_1^2 + S_2^2} \left[S_2(1-X) + ZS_1 \right] \right\}$$
 (37)

$$a_5 = \frac{2X \left[(1 - X)d_1 - Zd_2 \right]}{d_1^2 + d_2^2} = (1 + m_2^2 - m_1^2)$$
 (38)

$$b_5 = \frac{2X \left[Zd_1 + (1 - X)d_2 \right]}{d_1^2 + d_2^2} = 2m_2$$
 (39)

$$a_{6} = \frac{S_{1} \left\{ (Y \sin^{2} \psi)^{2} + 2 \cos^{2} \psi \left[(1 - X)^{2} - Z^{2} \right] \right\} - S_{2} \left[(2 \cos \psi)^{2} Z (1 - X) \right]}{S_{1}^{2} + S_{2}^{2}}$$
(40)

$$b_{6} = \frac{S_{2} \left\{ (Y \sin^{2} \psi)^{2} + 2 \cos^{2} \psi \left[(1 - X)^{2} - Z^{2} \right] \right\} + S_{1} \left[(2 \cos \psi)^{2} Z (1 - X) \right]}{S_{1}^{2} + S_{2}^{2}}$$
(41)

$$a_7 = \left\{ (2 - X) + \frac{2(Y \cos \psi)^2}{S_1^2 + S_2^2} \left[S_1(1 - X) - S_2 Z \right] \right\}$$
 (42)

$$b_7 = \left\langle 2Z + \frac{2(Y \cos \psi)^2}{s_1^2 + s_2^2} \left[s_1 Z + s_2 (1 - X) \right] \right\rangle$$
 (43)

$$a_8 = 2 \left\{ 1 + \frac{(Y \cos \psi)^2}{S_1^2 + S_2^2} \left[(1 - X^2)S_1 - S_2 Z(1 + X) \right] \right\}$$
 (44)

$$b_8 = \left\{ Z(2 + X) + \frac{2(Y \cos \psi)^2}{S_1^2 + S_2^2} \left[S_1 Z(1 + X) + S_2(1 - X^2) \right] \right\}$$
 (45)

Nomenclature	Used	in Ray	Trace	Equations

r, θ, φ	spatial coordinates of a spherical system
र्दे	wave normal or refractive index vector
$\sigma_{\mathbf{r}}$, $\sigma_{\boldsymbol{\theta}}$, $\sigma_{\boldsymbol{\phi}}$	vector components (ਰੋ)
τ	time of phase travel (in units of length)
μ	real part of complex phase refractive index
f	electromagnetic wave frequency
С	velocity of light in vacuum
m, e	mass and charge of an electron
Re	real part of the complex expression
Im ·	imaginary part of the complex expression
M	complex phase refractive index
x	scalar quantity representing normalized electron density
ω _p	plasma frequency at the spatial point
w	angular wave frequency
N	electron density at spatial point
Y	normalized magnitude of the earth's magnetic field vector $\overrightarrow{\overline{Y}}$
ω _c	magnitude of the gyromagnetic frequency of the electron about the earth's magnetic field
z	normalized collision frequency (v/ω)
٧	collision frequency at spatial point in collisions/sec
*	angle defined by inner product of vectors $\vec{\sigma}$ and \vec{Y}

*	imaginary part of the complex phase refractive index
k	absorption coefficient of the wave per unit length of path (proportional to the conductivity of the medium)
i	represents each of the spatial spherical coordinates $r,\;\theta,\phi$
Т	time of travel (seconds)
s	optical path length (km)
A	one-way absorption (nepers)
v _p	spatial velocity with which a point of constant phase moves
v _g	group velocity - spatial velocity of electromag- netic energy travel

C. COORDINATE TRANSFORMATION

If one takes three orthogonal planes intersecting at a point, one knows that the position of any point S in space is uniquely determined by the three perpendiculars from S on these planes, each with its proper sign. However, the problem of selecting the most useful orientation of such an orthogonal system is difficult since the usefulness of a coordinate system partially depends on the problem definition and the application of its solution. This ray trace program is designed as a sub-set of a much larger computer programming effort where the earth's geomagnetic field plays an important part. To minimize the number of computer transformations in the design of the over-all program, an earth centered spherical coordinate system (r, θ, φ) was chosen, whose z-component is coincident with the magnetic dipole axis.

This selection permits the application of the computer program to a great many studies of ray path problems because it accounts for the earth curvature and accepts for solution any electromagnetic radiating source whose transmitter location specifications of elevation, E, and azimuth, A, angles, as well as, geographic latitude, ϕ_R , geographic longitude, λ_R , and position with respect to the surface of the earth are known. Because the usefulness of this computer program can be extended by modification to other coordinate systems, as for example, an earth centered geographic system, or a radar coordinate system, the necessary coordinate transformations from the radar to the earth centered geomagnetic coordinate system will be described in detail.

For the discussion of this coordinate transformation, it is assumed that the electromagnetic wave transmitter is earth bound (i.e., fixed to the surface of the earth) at a geographic latitude ϕ_R and a geographic longitude λ_R . It is assumed that the radar is positioned so that the transmitting direction is described by the elevation angle, E, with respect to the tangent plane to the earth surface at the radar location, and an azimuth angle, A, measured from the radar coordinate that is tangent to a great circle passing through the north geographic pole. The azimuth angle is plus when measured counter-clockwise from the coordinate axis, ζ , whose positive direction points in the direction of geographic north. It is further assumed that S is a spatial point on the non-deviated portion of the ray a distance, R, from the electromagnetic wave transmitter. This is the spatial starting

point at which the numerical methods necessary for solution of the differential equations, must be initialized.

A transformation is required from the spherical radar coordinate system to the magnetic coordinate system whose origin is at the center of the earth.

The necessary matrices which are required to transform R, A, E coordinates to r, θ , ϕ coordinates can be arrived at by a series of simple matrix transformations.

1. Let ϵ , η , ζ be a set of orthogonal coordinates with origin on the surface of the earth. Let ϵ -axis be perpendicular to the earth's surface while ζ is directed (geographically) northward and η to the east. As stated above R is the slant range; E is the elevation angle; and A is the azimuth angle. Hence going from R, E, A to ϵ , η , ζ

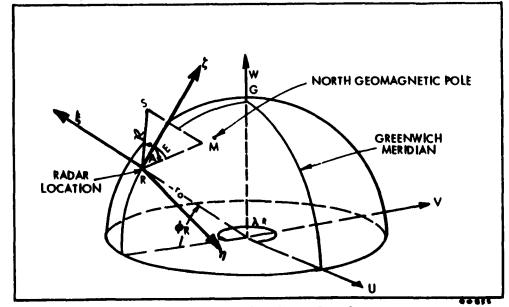


Figure 1a. Geometry of Coordinate Transformation

2. Let x_1 , y_1 , z_1 equal an orthogonal coordinate system with origin on the earth's axis of rotation. The x_1 -axis is in the latitude plane of the radar site and passes through the radar site. The z_1 -axis is coincident with the north geographic coordinate w. A translation and rotation is required in going from ε , η , ζ to x_1 , y_1 , z_1 .

$$\begin{pmatrix} \mathbf{x}_{1} \\ \mathbf{y}_{1} \\ \mathbf{z}_{1} \end{pmatrix} = \begin{pmatrix} \cos \varphi_{R} & 0 & -\sin \varphi_{R} \\ 0 & 1 & 0 \\ \sin \varphi_{R} & 0 & \cos \varphi_{R} \end{pmatrix} \begin{pmatrix} \varepsilon \\ \eta \\ \zeta \end{pmatrix} + \begin{pmatrix} \mathbf{r}_{o} \cos \varphi_{R} \\ 0 \\ 0 \end{pmatrix}$$

$$\begin{pmatrix} \mathbf{x}_{1} \\ \mathbf{y}_{1} \\ \mathbf{z}_{1} \end{pmatrix} = (\mathbf{b}_{ij}) \begin{pmatrix} \varepsilon \\ \eta \\ \zeta \end{pmatrix} + (\mathbf{b}_{i})$$

$$(47)$$

where r_{o} equals the earth's radius.

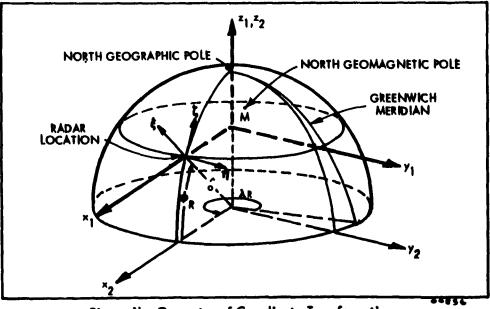


Figure 1b. Geometry of Coordinate Transformation

3. Let x_2 , y_2 , z_2 equal the orthogonal coordinate system with origin at the earth's center. Let the x_2 -axis be parallel to the x_1 -axis and z_2 coincide with w, hence also with z_1 . Then going from x_1 , y_1 , z_1 to x_2 , y_2 , z_2 by translation

$$\begin{pmatrix} x_2 \\ y_2 \\ z_2 \end{pmatrix} = \begin{pmatrix} x_1 \\ y_1 \\ z_1 \end{pmatrix} + \begin{pmatrix} 0 \\ 0 \\ r_0 \sin \varphi_R \end{pmatrix} = \begin{pmatrix} x_1 \\ y_1 \\ z_1 \end{pmatrix} + (c_i)$$
(48)

4. Let x_3 , y_3 , z_3 represent an orthogonal coordinate system with origin at the earth's center such that the x_3 -axis intersects the zero degree longitudinal geomagnetic meridian while the z_3 -axis coincides with w. Hence going from x_2 , y_2 , z_2 to x_3 , y_3 , z_3 by rotation about the z_2 -axis yields,

$$\begin{pmatrix} \mathbf{x}_{3} \\ \mathbf{y}_{3} \\ \mathbf{z}_{3} \end{pmatrix} = \begin{pmatrix} \cos(\lambda_{\mathbf{M}} - \lambda_{\mathbf{R}}) & \sin(\lambda_{\mathbf{M}} - \lambda_{\mathbf{R}}) & 0 \\ -\sin(\lambda_{\mathbf{M}} - \lambda_{\mathbf{R}}) & \cos(\lambda_{\mathbf{M}} - \lambda_{\mathbf{R}}) & 0 \\ 0 & 0 & 1 \end{pmatrix} \begin{pmatrix} \mathbf{x}_{2} \\ \mathbf{y}_{2} \\ \mathbf{z}_{2} \end{pmatrix}$$

$$\begin{pmatrix} x_3 \\ y_3 \\ z_3 \end{pmatrix} = \begin{pmatrix} d_{ij} \end{pmatrix} \begin{pmatrix} x_2 \\ y_2 \\ z_2 \end{pmatrix}$$
(49)

where as before λ_{M} and Ψ_{M} represent the geographic longitude and latitude of the geomagnetic north pole M.

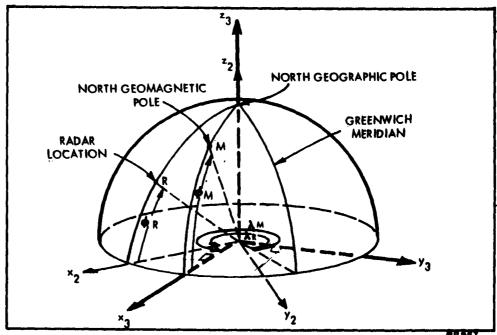


Figure 1c. Geometry of Coordinate Transformation

5. Let x, y, z represent an orthogonal coordinate system with origin at the earth's center. Let the x-axis pass through the great circle connecting the geographic and geomagnetic poles while the z-axis passes through the geomagnetic pole M. In going from x_3 , y_3 , z_3 to x, y, z by a rotation one obtains

$$\begin{pmatrix} \mathbf{x} \\ \mathbf{y} \\ \mathbf{z} \end{pmatrix} = \begin{pmatrix} \sin \varphi_{\mathbf{M}} & 0 & -\cos \varphi_{\mathbf{M}} \\ 0 & 1 & 0 \\ \cos \varphi_{\mathbf{M}} & 0 & \sin \varphi_{\mathbf{M}} \end{pmatrix} \begin{pmatrix} \mathbf{x}_{3} \\ \mathbf{y}_{3} \\ \mathbf{z}_{3} \end{pmatrix}$$

$$\begin{pmatrix} x \\ y \\ z \end{pmatrix} = (e_{ij}) \begin{pmatrix} x_3 \\ y_3 \\ z_3 \end{pmatrix}$$
 (50)

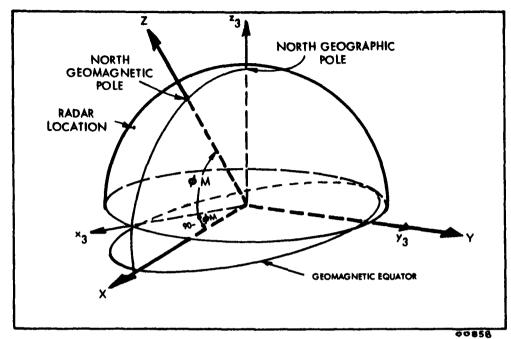


Figure 1d. Geometry of Coordinate Transformation

Hence by matrix multiplication one can transform from radar coordinates R, E, A to the earth centered coordinates x, y, z where the magnetic dipole axis of the earth coincides with the z-axis. From this coordinate system one can simply transform to the desired spherical coordinate system r, θ , Φ .

$$\begin{pmatrix} \mathbf{x} \\ \mathbf{y} \\ \mathbf{z} \end{pmatrix} = (\mathbf{e}_{ij}) \begin{pmatrix} \mathbf{x}_3 \\ \mathbf{y}_3 \\ \mathbf{z}_3 \end{pmatrix} = (\mathbf{e}_{ij}) (\mathbf{d}_{ij}) \begin{pmatrix} \mathbf{x}_2 \\ \mathbf{y}_2 \\ \mathbf{z}_2 \end{pmatrix} = (\mathbf{e}_{ij}) (\mathbf{d}_{ij}) \begin{pmatrix} \mathbf{x}_1 \\ \mathbf{y}_1 \\ \mathbf{z}_1 \end{pmatrix} + (\mathbf{c}_i)$$

$$= (\mathbf{e}_{ij}) (\mathbf{d}_{ij}) \begin{pmatrix} \mathbf{\epsilon} \\ \mathbf{\eta} \\ \mathbf{\zeta} \end{pmatrix} + (\mathbf{b}_i + \mathbf{c}_i)$$

$$(51)$$

$$(g_{ij}) = (e_{ij}) (d_{ij}) (b_{ij}) = (f_{ij}) (b_{ij})$$
 (52)

$$(f_{ij}) = \begin{pmatrix} \sin \varphi_{M} \cos(\lambda_{M} - \lambda_{R}) & \sin \varphi_{M} \sin(\lambda_{M} - \lambda_{R}) & -\cos \varphi_{M} \\ -\sin(\lambda_{M} - \lambda_{R}) & \cos(\lambda_{M} - \lambda_{R}) & 0 \\ \cos \varphi_{M} \cos(\lambda_{M} - \lambda_{R}) & \cos \varphi_{M} \sin(\lambda_{M} - \lambda_{R}) & \sin \varphi_{M} \end{pmatrix}$$
(53)

$$g_{ij} = \begin{pmatrix} g_{11} & g_{12} & g_{13} \\ g_{21} & g_{22} & g_{23} \\ g_{31} & g_{32} & g_{33} \end{pmatrix}$$
(54)

$$g_{11} = \left[\cos \varphi_{R} \sin \varphi_{M} \cos(\lambda_{M} - \lambda_{R}) - \cos \varphi_{M} \sin \varphi_{R}\right]$$

$$g_{12} = \sin \varphi_{M} \sin(\lambda_{M} - \lambda_{R})$$

$$g_{13} = \left[-\sin \varphi_{R} \sin \varphi_{M} \cos(\lambda_{M} - \lambda_{R}) - \cos \varphi_{R} \cos \varphi_{M}\right]$$

$$g_{21} = -\sin(\lambda_{M} - \lambda_{R}) \cos \varphi_{R}$$

$$g_{22} = \cos(\lambda_{M} - \lambda_{R})$$

$$g_{23} = \sin \varphi_{R} \sin(\lambda_{M} - \lambda_{R})$$

$$g_{31} = \left[\cos \varphi_{R} \cos \varphi_{M} \cos(\lambda_{M} - \lambda_{R}) + \sin \varphi_{M} \sin \varphi_{R}\right]$$

$$g_{32} = \cos \varphi_{M} \sin(\lambda_{M} - \lambda_{R})$$

$$g_{33} = \left[-\sin \varphi_{R} \cos \varphi_{M} \cos(\lambda_{M} - \lambda_{R}) + \sin \varphi_{M} \cos \varphi_{R}\right]$$

$$\begin{pmatrix} x \\ y \\ z \end{pmatrix} = R (g_{ij}) \begin{pmatrix} \sin E \\ \cos E \sin A \\ \cos E \cos A \end{pmatrix} + r_{o} (g_{i1}) = r \begin{pmatrix} \sin \theta \cos \phi \\ \sin \theta \sin \phi \\ \cos \theta \end{pmatrix}$$
(55)

By use of these matrix transformations the Cartesian coordinates, (x, y, z), and from them the spherical coordinates, (r, θ, ϕ) , of the earth centered geomagnetic coordinate system can be determined for the spatial starting point S and the earth bound transmitter R. These can be expressed as:

$$\mathbf{x}_{S} = \left[\cos(\lambda_{M} - \lambda_{R}) \sin\phi_{M} \cos\phi_{R} - \cos\phi_{M} \sin\phi_{R}\right] \quad (R \sin E + r_{o})$$

$$+ \left[\sin(\lambda_{M} - \lambda_{R}) \sin\phi_{M}\right] \quad R \cos E \sin A \qquad (56)$$

$$- \left[\cos(\lambda_{M} - \lambda_{R}) \sin\phi_{M} \sin\phi_{R} + \cos\phi_{M} \cos\phi_{R}\right] \quad R \cos E \cos A$$

$$y_{S} = \left[-\sin(\lambda_{M} - \lambda_{R})\cos\phi_{R}\right] \left(R\sin E + r_{o}\right) + \left[\cos(\lambda_{M} - \lambda_{R})\right] R\cos E \sin A$$

$$+ \left[\sin(\lambda_{M} - \lambda_{R})\sin\phi_{R}\right] R\cos E \cos A \qquad (57)$$

$$z_{S} = \left[\cos(\lambda_{M} - \lambda_{R})\cos\phi_{M}\cos\phi_{R} + \sin\phi_{M}\sin\phi_{R}\right] \quad (R \sin E + r_{o})$$

$$+ \left[\sin(\lambda_{M} - \lambda_{R})\cos\phi_{M}\right] \cdot R \cos E \sin A \qquad (58)$$

$$- \left[\cos(\lambda_{M} - \lambda_{R})\cos\phi_{M}\sin\phi_{R} - \sin\phi_{M}\cos\phi_{R}\right] \cdot R \cos E \cos A$$

When R = 0, that is, for a point on the surface of the earth,

$$\mathbf{x}_{\mathbf{R}} = \mathbf{r}_{\mathbf{O}} \left[\cos(\lambda_{\mathbf{M}} - \lambda_{\mathbf{R}}) \sin \phi_{\mathbf{M}} \cos \phi_{\mathbf{R}} - \cos \phi_{\mathbf{M}} \sin \phi_{\mathbf{R}} \right]$$
 (59)

$$y_{R} = -r_{0} \sin(\lambda_{M} - \lambda_{R}) \cos \varphi_{R}$$
 (60)

$$z_{R} = r_{o} \left[\cos(\lambda_{M} - \lambda_{R}) \cos\phi_{M} \cos\phi_{R} + \sin\phi_{M} \sin\phi_{R} \right]$$
 (61)

From simple trigonometric considerations (Figure 2a) it can be shown that the radar slant range, R, measured from the transmitter to the spatial starting point S is given by

$$R = -r_{o} \sin E + \sqrt{(r_{o} + h_{S})^{2} - r_{o}^{2} \cos^{2}E}$$
 (62)

where h_S is the vertical height of the starting point above its projection, (point P) on the surface of the earth.

Equations 1 through 6 point out that in addition to these transformations, the components of the directed normal to the phase fronts, $\vec{\sigma}$, at the starting point S are to be determined in this coordinate system. From spherical and plane trigonometric considerations, (Figure 2), it can be shown that these components are given by

$$\sigma_r = \sigma \cos e$$
 (63)

$$\sigma_{\Omega} = \sigma \sin e \cos \alpha$$
 (64)

$$\sigma_{\varphi} = -\sigma \sin \alpha \tag{65}$$

Angle e can be evaluated directly by employing the law of sines. This yields

$$e = \sin^{-1} \left(\frac{r_0 \cos E}{r_0 + h_S} \right)$$
 (66)

Angle a is the geomagnetic bearing angle (Figure 2) measured positive in a clockwise direction from geomagnetic north. By use of spherical trigonometry it is expressible by

$$a = \tan^{-1} \left[\frac{\sin(\bar{\Phi}_{S} - \bar{\Phi}_{R})\sin\theta}{\cos\theta_{S}\sin\theta_{R}\cos(\bar{\Phi}_{S} - \bar{\Phi}_{R}) - \sin\theta_{S}\cos\theta_{R}} \right]$$
(67)

where Φ_R , θ_R , and Φ_S , θ_S are the geomagnetic longitudes and colatitudes, respectively, of the radar transmitter, R, and the spatial starting point, S. The geomagnetic angles are obtained from the following expressions

$$\Phi_{R} = \tan^{-1} \left(\frac{y_{R}}{x_{R}} \right) = \tan^{-1} \left(\frac{g_{21}}{g_{11}} \right)$$

$$= \tan^{-1} \left[\frac{-\sin(\lambda_{M} - \lambda_{R})\cos\phi_{R}}{\cos(\lambda_{M} - \lambda_{R})\sin\phi_{M}\cos\phi_{R} - \cos\phi_{M}\sin\phi_{R}} \right] (68)$$

$$\theta_{R} = \cos^{-1} \left(\frac{z_{R}}{r_{R}} \right) = \cos^{-1} (g_{31})$$

$$= \cos^{-1} \left[\cos(\lambda_{M} - \lambda_{R}) \cos \phi_{M} \cos \phi_{R} + \sin \phi_{M} \sin \phi_{R} \right]$$
 (69)

$$\Phi_{S} = \tan^{-1}\left(\frac{y_{S}}{x_{S}}\right) \tag{70}$$

$$\theta_{S} = \cos^{-1} \frac{z_{S}}{r_{o} + h_{S}} \tag{71}$$

One additional useful expression can be obtained from these algebraic relations. The parameter is the angle \forall at the spatial starting point S which is defined by the inner product of the magnetic field vector and the wave normal. By the application of the sine and cosine laws to the geometry of Figure 2-a, it can be shown that

$$\Psi = \cos^{-1} \left[-(\cos e \sin I + \sin e \cos I \cos a) \right]$$
 (72)

where angle I is the magnetic inclination angle. The inclination angle 9 is only a function of the geomagnetic latitude at the particular point in question.

$$I = \tan^{-1} \left[2 \cot \theta_{S} \right]$$
 (73)

Expressions arising from the inverse coordinate transformation, that is, transformation from the geomagnetic coordinates (r, θ, ϕ) to the radar coordinates R, E, A can be easily developed from these formulae. Although used in the computer program they will not be presented here.

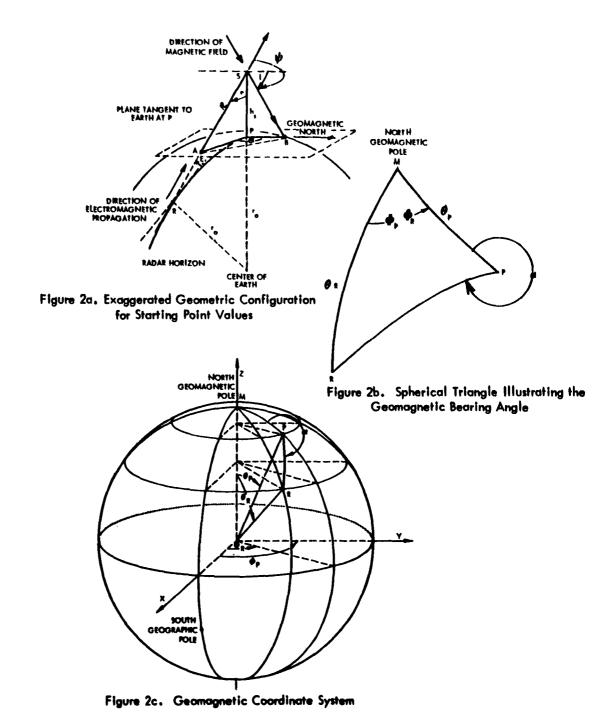


Figure 2. Geometry for Starting Point and Geomagnetic Coordinate System

r, θ,φ	spatial point in an earth centered spherical co- ordinate system
E	radar elevation angle
A	radar azimuth angle
^φ R, M	geographic latitude of point R or M, respectively
λ _{R, M}	geographic longitude of point R or M, respectively
S	spatial starting point on nondeviated portion of ray
R	distance from electromagnetic wave transmitter to starting point
ε, η,ζ	set of orthogonal coordinates with origin on the surface of the earth at radar site
x ₁ , y ₁ , z ₁	orthogonal coordinate system with origin on the earth's axis of rotation
*2', y2', z	orthogonal coordinate system with origin at earth's center
*3', y3', z3	orthogonal coordinate system with origin at earth's center
w	z component of the geographic coordinate system (u, v, w)
^h S	height of starting point above its projection on the surface of the earth
ro	radius of earth
α	geomagnetic bearing angle
Φ _R , Φ _S	geomagnetic longitudes of points R and S

RM 61TMP-32

θ_{R}, θ_{S}	geomagnetic co-latitudes of points R and S
σ _r , σ _θ ,σ _φ	physical components of a vector of length μ , that is directed normal to the phase front
Ψ	angle between magnetic field vector and the wave normal
I	angle of magnetic inclination

D. MODEL IONOSPHERE

As shown under Computational Procedure, the refractive index M and its spatial derivatives are dependent on the normalized density, X, and its spatial gradients, $\partial X/\partial r$, $\partial X/\partial \theta$, and $\partial X/\partial \phi$. For an evaluation of these quantities an analytic model ionosphere can be chosen. One such ionospheric model that was found useful, is a spherical electron distribution as measured from a point in space. A reason for its selection is presented under Computational Results. Other uses as well as other ionospheric models are covered elsewhere 7, 11.

Let r_b , θ_b , ϕ_b represent the spatial location, B, of the center of the selected spherical ionosphere in the geomagnetic spherical coordinate system (r, θ, ϕ) . The values of these coordinates are obtainable from the specified geographic latitude, longitude and height above the earth surface of point B, in an analogous procedure as described for the transformation of coordinates of the spatial starting point S. Then the electron density at a spatial point r, θ , ϕ for an assumed spherical ionosphere can be written as:

N(r,
$$\theta$$
, Ψ , r_b , θ_b , Ψ_b) = $\frac{A}{\mathcal{C}^n}$ (74)

where

$$\mathcal{L} = \left\{ \left(\mathbf{r}_{b} \sin \theta_{b} \cos \phi_{b} - \mathbf{r} \sin \theta \cos \phi \right)^{2} + \left(\mathbf{r}_{b} \sin \theta_{b} \sin \phi_{b} - \mathbf{r} \sin \theta \sin \phi \right)^{2} + \left(\mathbf{r}_{b} \cos \theta_{b} - \mathbf{r} \cos \theta \right)^{2} \right\}^{1/2} = \left\{ \mathbf{x} \mathbf{P}^{2} + \mathbf{y} \mathbf{P}^{2} + \mathbf{z} \mathbf{P}^{2} \right\}$$
(75)

For this discussion A and n are appropriately chosen constants which give the desired electron density N (electrons/cc). By use of Equations 74 and 75, it is easily shown that the spatial electron density gradients can be expressed in the following manner,

$$\frac{\partial N}{\partial r} = \frac{nA}{\varphi^{n+2}} \left[XP \sin \theta \cos \varphi + YP \sin \theta \sin \varphi + ZP \cos \theta \right]$$
 (76)

$$\frac{\partial N}{\partial \theta} = \frac{nAr}{\rho^{n+2}} \left[XP \cos \theta \cos \phi + YP \cos \theta \sin \phi - ZP \sin \theta \right]$$
 (77)

$$\frac{\partial N}{\partial \varphi} = \frac{nAr}{e^{n+2}} \left[-XP \sin \theta \sin \varphi + YP \sin \theta \cos \varphi \right]$$
 (78)

Some results obtained by use of such a spherical ionospheric model will be discussed later.

E. MODEL OF EARTH'S MAGNETIC FIELD

Because magneto-ionic effects on the propagation of electromagnetic waves through an ionized medium are taken into account in the derivation of the equations under Computational Procedure, it is necessary to specify the normalized external magnetic field of the earth, \vec{Y} , its components Y_r , Y_θ , Y_θ and its spatial derivatives $\partial Y/\partial r$, $\partial Y/\partial \theta$, $\partial Y/\partial \varphi$. It is known that the earth's magnetic field can be approximated by an earth centered magnetic dipole with its axis displaced such that the geographic longitude $\lambda_M = 70.1^\circ W$ and the geographic latitude $\bar{\Phi}_M = 78.6^\circ N$. The magnetic potential, V, at a distant point from such a dipole is related to the magnetic moment, \mathcal{M} , by the expression

$$V(r,\theta) = -\frac{M\cos\theta}{r^2} = -\frac{(Y_e r_o^3)\cos\theta}{r^2}$$
 (79)

where r, θ are the geomagnetic coordinates of the spatial point irrespective of the coordinate Ψ and as before, r_0 = radius of the earth. In this equation Y_e is the magnitude of the normalized magnetic field at the earth's surface on the magnetic equator. By use of this algebraic equation all the desired quantities can be derived. They are

$$Y = Y_e \left(\frac{r_o}{r}\right)^3 (1 + 3 \cos^2 \theta)^{1/2}$$
 (80)

where, as previously defined, Y is the normalized magnitude of the earth's magnetic field vector \vec{Y} = (eH/mc ω) = ω_c/ω

$$Y_{r} = 2 Y_{e} \left(\frac{r_{o}}{r}\right)^{3} \cos \theta = \frac{Y}{\sqrt{1 + \frac{1}{4} \tan^{2} \theta}}$$
 (81)

$$Y_{\theta} = Y_{e} \left(\frac{r_{o}}{r}\right)^{3} \sin \theta = \frac{1}{2} Y_{r} \tan \theta$$
 (82)

$$Y_{\varphi} = \frac{\partial Y}{\partial \varphi} = 0 \tag{83}$$

$$\frac{\partial Y}{\partial r} = -\frac{3Y}{r} \tag{84}$$

$$\frac{\partial Y}{\partial \theta} = -\frac{3Y \sin \theta \cos \theta}{\left[1 + 3 \cos^2 \theta\right]} \tag{85}$$

F. MODEL OF ATMOSPHERIC COLLISION FREQUENCY

For some of the trial calculations the atmospheric collision frequency was found from assumed exponential variations of collision frequency with height. The atmosphere was radially stratified and an approximate exponential equation was curve-fitted to measured experimental data for each stratified region. Hence, for each region the following relations were used to obtain Z and $\delta Z/\delta r$, $\delta Z/\delta \theta$, $\delta Z/\delta \phi$ that are required by the ray trace equations.

$$Z = \frac{v}{\omega} = ae^{-b(r - r_0)}$$
 (86)

$$\frac{\partial Z}{\partial r} = -bZ \tag{87}$$

$$\frac{\partial Z}{\partial \theta} = \frac{\partial Z}{\partial \Psi} = 0 \tag{88}$$

The dependence of collision frequency on a localized temperature distribution and degree of ionization⁷ complicates these simple relations. These complications (as derived by D. Archer) as well as their effects will not be discussed at this time.

G. COMPUTATIONAL RESULTS

The preceding equations are only a summary of the required set which will permit the detailed calculation of a ray path in three-dimensional space. Because of this, it becomes clear that the only realistic approach to the solution of this problem is the utilization of computer techniques, otherwise, the welter of data that must be handled through use of numerical methods, is beyond effective human handling capacity. However, the development of a computer program which can perform countless number of calculations, poses the very difficult task of determining the correctness of a computed result. To simplify this "debugging" task the classical idea of elastic collision between charged particles was borrowed from nuclear physics. It has been shown 10 that if the electron density falls off as the inverse square of the distance from the center of a spherical electron distribution (N = A/ \mathcal{R}^2), various exact expressions can be obtained, since the ray equations at zero azimuth are integrable.

The geometry of such a distribution, as well as, three ray paths computed by use of the computer program are shown in Figure 3. In the figure the center of the sphere is located by the fixed coordinates (R_0, β_0) with respect to the radar. The derivations are made in two dimensions, hence only the two-dimensional coordinate system (ξ, ζ) is used. The ray path has an initial elevation angle E_1 and its distance of closest approach to the center of the refracting sphere is denoted by \mathcal{K}_0 . The coordinates of any point on the ray path with respect to the center of the sphere are (\mathcal{K}, β) and with respect to the radar (R, E). Angle δ represents the amount of ray bending experienced by the ray passing through the refractive medium. Under these conditions Archer 10 has shown that the angle δ is given by

$$\delta = \pi - 2 \left[\gamma_o + \mu \left(\mathcal{X}_o \right) \cos^{-1} \left(\frac{\mathcal{X}_o}{R_o} \right) \right]$$
 (89)

The three plotted rays have actually a small third dimensional component. Because of this, the accuracy of these plotted rays is approximately (+3, -2) percent.

As shown by the tabulated results in Table 1, the computed deviation angle δ_c , arising from ray trace results, agrees very well with the calculated angle, δ , obtained by use of Equation 89. These computer

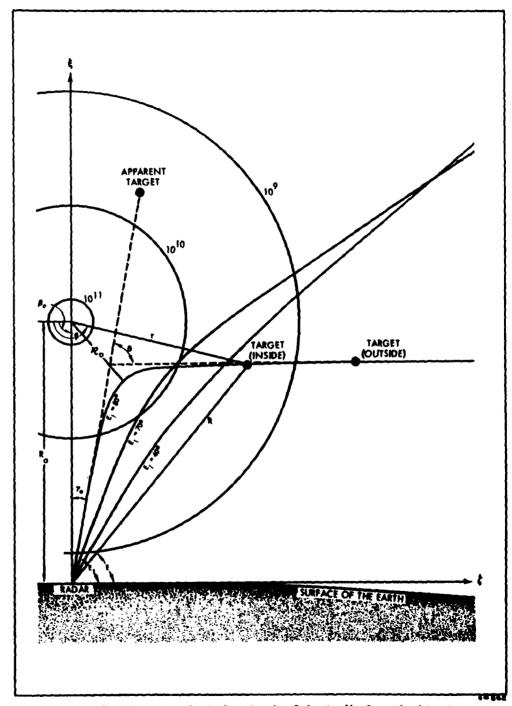


Figure 3. Geometry for Refraction by Spherically Stratified Region

calculations were performed under a large error upper bound condition ($E = 10^{-3}$, see description of Subroutine INT). The agreement is improved by a variation of this error condition. Additional results will not be presented here. Presented elsewhere 12 is the influence of E on computed results using these numerical methods as applied to the study of ionization-deionization phenomena.

Radar Elevation Angle - E Degrees	Bending Angle - 8 from Equation 89 Degrees	Bending Angle - δ _C Ray Trace Program Degrees
60	19. 5	20
70	35. 2	37
80	77.5	78

Table 1

Comparison of Total Ray Bending Angle

As an additional illustration, refractive errors through a particular spherical ionized region have been computed in detail to illustrate the concepts discussed and to indicate the kinds of refractive errors that could arise. The electron density contours of this ionization model are defined by Equation 74 where $A = 10^{33}$ and n = 12. The distance from the center of the spherical ionosphere is measured in kilometers. The center of the model is located at an elevation angle, E, of 30 degrees, zero degree azimuth angle, and 564 kilometers slant range as measured from the radar site.

Figure 4 shows the relation between the radar and ionization model in the plane of zero azimuth. Also shown are the ray paths for rays leaving the radar at several elevation angles. A frequency of one kilomegacycle was used in determining the refraction of the electromagnetic wave propagation vector. Because the electron density increases rapidly near the center of the model, there is significant bending of the ray path.

The refraction becomes so severe as the elevation angle of radar rays approach the elevation angle of the ionization center with respect to the radar, that there is a region (shown by half tones) into which no radar ray penetrates, hence, radar "blackout" is achieved. In three dimensions this blackout region is a cone in which a target is shielded from the radar. Because rays near this region intersect each other, two elevation angle paths to the same target exist, so multiple targets may be visible.

If the ray path is not in the zero azimuth plane, the amount of elevation error, or azimuth error, is a function of the location of the target. The elevation and azimuth errors for a target located at a slant range of 1200 kilometers have been computed as a function of radar elevation and azimuth angles. These are summarized in Figure 5, in which contours of constant elevation error, ΔE , in one quadrant and constant azimuth error, ΔA , in another quadrant as a function of the ray direction at the radar site are given. The contours have been terminated at a total bearing error of about 10 degrees. Due to symmetry the errors in the other quadrants are just the mirror image of the quadrants shown.

RM 61TMP-32

Nomenclature Used for Computational Results

E	initial elevation angle made by ray path
\mathscr{L}_{\circ}	distance of closest approach between ray path and center of sphere
(L, B)	coordinates of any point on the ray path with respect to the center of the sphere
(R, E)	coordinates of any point on the ray path with respect to the radar
Ro	distance between the radar and center of the sphere
Yo	apparent bearing angle

SECTION III

COMPUTER PROGRAM FOR THREE-DIMENSIONAL RAY-TRACING

Figure 6 schematically describes the computer program that was developed for three-dimensional ray tracing. As illustrated, the computer program is a composite of a group of subprograms. Because each subroutine is an entity in itself, the improvement of the entire program can be performed by the variation of each subprogram.

For the creation of this program the FORTRAN language 13 was used wherever possible. FORTRAN is an automatic coding.system for the IBM-704/709/7090 Data Processing Computer System that was designed for scientific application. Although there are limitations to FORTRAN, nevertheless, 1) it is at present the only language for scientific use, that is accepted by most existing large computer systems, and 2) it is simple and therefore without much effort, permits the elimination of the programmer, thus leaving the design of logical computer decisions, to the formulator of the scientific problem. The program has been written to operate "in or out" of the FORTRAN MONITOR CONTROL SYSTEM.

Except for the RINDEX subroutine the program has been divided into small, simple Functions and Subroutines to facilitate understanding. In the development of the program, concentration was mainly on obtaining a correct working program, as soon as possible, and not on optimization or clarity of output results. These tasks are left for future development.

The computer program consists of the following parts:

1)	Main Program	RAY TRACE
2)	Function	SLANTR
3)	Function	QATAN
4)	Function	ARCOS

5)	Subroutine	COORD
6)	Subroutine	DAUX
7)	Subroutines	INT and INTM
8)	Subroutine	RINDEX
9)	Subroutine	ELECTX
10)	Subroutine	BIGR
11)	Subroutine	MAGY
12)	Subroutine	COLFRZ
13)	Subroutine	RCOORD
14)	Subroutine	OUTONE

15) Subroutine

These functions and subroutines are used to obtain numerical values for those variables which cannot be defined by only one arithmetic statement. In addition to these subprograms certain statements in the FORTRAN language cause the inclusion in the object program of the necessary input and output routines, as well as, various library functions and subroutines in relocatable binary form that are available on the FORTRAN MASTER LIBRARY TAPE. The names and locations of these necessary routines are given in the "storage map" of the arrangement of storage location in the object program that is compiled from a FORTRAN source program. These "maps" follow the listings of each source program. These added routines will not be discussed in this report. Following a brief description of the function of the main program and its associated subprograms, the necessary input data for a sample calculation is given with a description of the output. Some of the results listed in this output led to the graphical results presented in Figures 3, 4 and 5.

OUTPUT

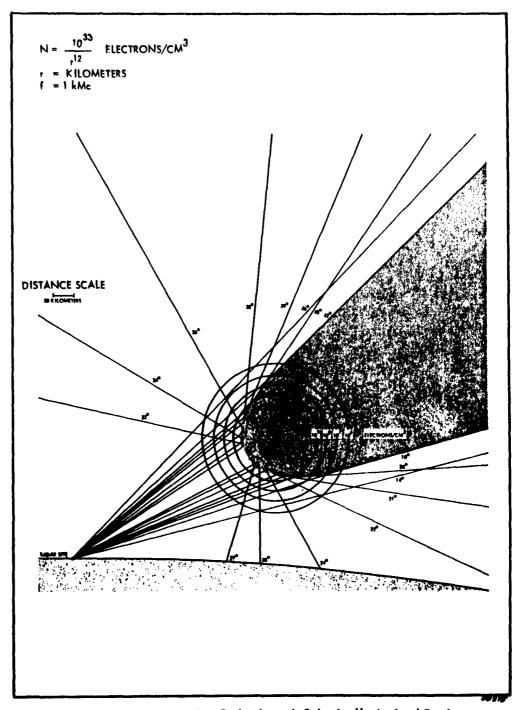


Figure 4. Radar Propagation Paths through Spherically Ionized Region

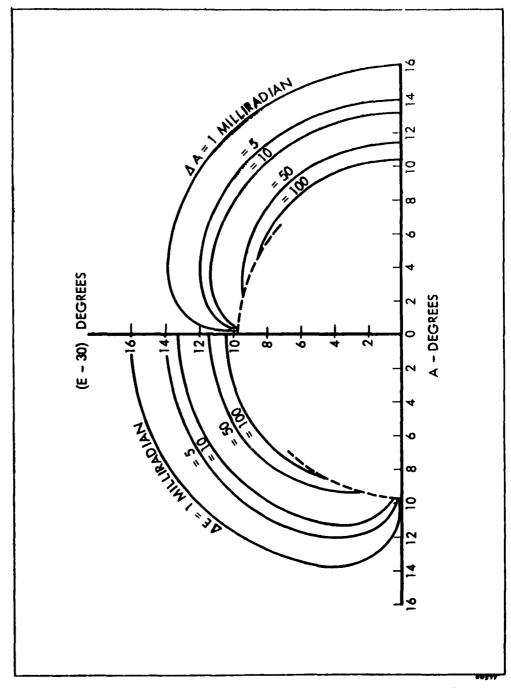


Figure 5. Elevation and Azimuth Errors for Propagation through a Spherical Model – $f=1~\mathrm{kMc}$

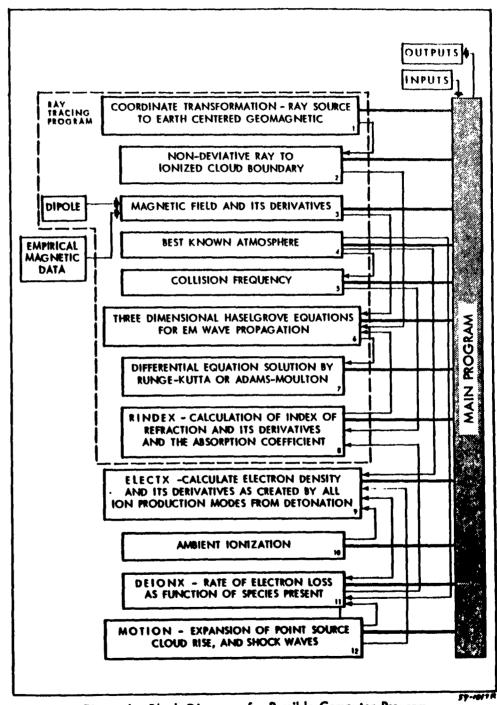


Figure 6. Block Diagram of a Possible Computer Program

A. MAIN PROGRAM RAY TRACE

The main program's function is to act as a master control of the logical flow necessary for the execution of the numerical methods. In addition, it is responsible for obtaining the necessary data, initializing the required starting conditions, performing the desired controlled printouts of computed results, and determining the condition for termination of the given computations. For initializing the starting conditions the main program requires the reading into storage of the following information in the format illustrated under INPUT, Table 4.

CARD 1

RECORD: This can be any desired information consisting of 72 alphanumeric characters that will serve to identify the calculations.

CARD 2

Contains the values of ID, KWIT. ID is an integer that can be used for identifying the calculation if the same CARD 1 is used. KWIT: On completion of the calculations in progress, the computer will check if additional problems are to be performed. Thus if KWIT=59 the computer will want to read a new W vector (CARD 3 --- onward); if KWIT=66, the computer will want to read a new CARD 1, CARD 2, and new W vector (CARD 3 --- onward); if KWIT equals any other integer the computer will PAUSE 44444.

CARD 3 onward

This card and all following cards describe the value of each component of the W vector that is not zero. As shown under INPUT, the first three columns of the card are for the integer that describes the W vector component. The next fourteen columns of the card are for the value of the W vector component. The number of these cards is variable since on completion of one calculation, often only one component of the W vector

RM 61TMP-32

CARD 3 (continued)

is to be changed for the next computation. The W vector can be read in any order.

LAST DATA CARD

This card follows the last card describing the W vector. It is any negative integer listed in the first three columns o a card. It transfers the computer out of the read mode to the location beginning the ray trace calculations.

SENSE SWITCH 1

The program is designed to calculate first the ordinary ray path and then the extraordinary ray path. SENSE SWITCH 1 DOWN will eliminate the calculation of the extraordinary ray path.

SENSE SWITCH 2

In DOWN position will permit the calculation of the extraordinary ray path and eliminate the calculation of the ordinary ray path.

SENSE SWITCH 3

Placing this SENSE SWITCH 3 DOWN will terminate the calculation on completion of the ray path calculations in progress.

SENSE SWITCH 4

Placing SENSE SWITCH 4 DOWN will cause the computer to check if SENSE SWITCH 6 is DOWN. If it is down the computer will terminate calculations immediately.

SENSE SWITCH 6

It is desirable to follow the course of any calculation on a computer. SENSE SWITCH 6 DOWN will print on-line, the total number of numerical integrations completed up to this point, integration mesh size, length of independent variable τ , height above surface of the earth (km), θ , Ψ (in degrees), σ_r , σ_θ , σ_{φ} , μ , π , distance from the ion source center (km), value of the normalized electron density X.

SENSE LIGHT 2

When SENSE SWITCHES 1 and 2 are DOWN, then both ordinary and extraordinary ray paths are being calculated. When SENSE LIGHT 2 is ON, then the calculation is determining the ordinary ray path. When it is OFF, then the extraordinary ray path is being calculated.

PAUSE 17171

If the computer halts with this octal number in the address field of the STORAGE REGISTER it signifies that SENSE SWITCHES 1 and 2 are in the UP position and the problem is undefined. SENSE SWITCHES 1 or 2, or 1 and 2 are to be placed DOWN depending if only the ordinary, the extraordinary, or both ray paths are to be calculated. Following the definition of the problem, pushing START key will cause calculations to resume.

PAUSE 66666

The computer halts with this octal number in the STORAGE REGISTER just prior to beginning calculations. If the MONITOR system is used it permits the operator to know when it has left the MONITOR system and the SENSE SWITCHES can be changed as needed by the problem.

PAUSE 44444

The computer halts with this octal number in the STORAGE REGISTER on completion of all the necessary calculations specified by the INPUT data. It permits the operator to reset the desired sense switches for the MONITOR system. Pressing START will cause the computer to exit from the program to the MONITOR system.

Table 2 contains the nomenclature that describes some of the components of the V vector and the components of the W vector.

V(2)	independent variable τ
V(3)	initial step size input $\Delta \tau$
V(4)	radius from center of earth r
V(5)	variable angle θ
V(6)	variable angle Φ
V(7)	$\sigma_{\mathbf{r}}$
V(8)	σ _θ
V(9)	$\sigma_{oldsymbol{\phi}}$
V(10)	optical path length one way s
V(11)	time one way T
V(12)	A absorption
V(13)	dr/dŢ
V(14)	dθ/dτ
V(15)	dφ/dτ
V(16)	dσ _r /dτ
V(17)	dσ _θ /dτ
V(18)	dσ _φ /dτ
V(19)	ds/dT
V(20)	dT/dT
V(21)	dA/dT

Table 2. Nomenclature Describing the V and W Vectors. (Page 1 of 6)

W(1)	refractive index μ
W(2)	imaginary part of complex phase refractive index *
W(3)	radar transmitter angular frequency ω
W (4)	ôμ/ôσ _r
W (5)	ðμ/ðσ _θ
W(6)	дн /дα ^ф
W (7)	ðµ/ðr
W (8)	θμ/θθ
W (9)	ФФ/ 46
W(10)	a⊬/a ∜
W(11)	ω6\μ6
W(12)	unassigned for this program
W(13)	geographic longitudinal angle $\lambda_{\mathbf{M}}$ of geomagnetic north pole measured east of Greenwich Meridian (degrees)
W(14)	angle λ_R measured as W(13) in degrees
W(15)	angle Ψ_{M} geographic latitude of geomagnetic northpole measured plus from geographic equator north
W(16)	angle φ_R geographic latitude of radar (degrees)
W(17)	radar elevation angle E (degrees)
W(18)	radar azimuth bearing angle angle A (degrees)
W(19)	r _o radius of the earth (km)
W(20)	hs height of starting point above surface of earth
	Table 2. Nomenclature Describing the V and W Vectors. (Page 2 of 6)

W(21)	angle $\phi_{\mbox{\footnotesize B}}$ of ionization source measured as W(15) (degrees)
W (22)	longitudinal angle λ_{B} of source measured as W(13)
W (23)	$\mathbf{h}_{\mathbf{B}}$ height of ionization source center above earth surface
W (24)	Δau initial mesh size of variable
W (25)	$\mathbf{Y}_{\mathbf{e}}$ normalized equator magnetic field on earth's surface at the geomagnetic equator
W (26)	a constant determining collision frequency
W (27)	b constant in exponent determining collision frequency
W (28)	range (km) = distance from ionization source center to spatial point (r, θ, φ) =
W(29)	cosine of angle makes with the vertical through center of the ionizing source
W(30)	R _b radial distance from earth's center to center of ionizing source
W(31)	x geomagnetic coordinate of source
W(32)	y geomagnetic coordinate of source
W(33)	z geomagnetic coordinate of source
W(34)	A = constant in A/Rn determining electron density
W (35)	$n = exponent in A/R^n$ equation
W(36)	unassigned for this program
W(37)	unassigned for this program
W (38)	plasma angular frequency cycles/sec

Table 2. Nomenclature Describing the V and W Vectors. (Page 3 of 6)

1	W(39)	Ne in ion pairs/cc
1	W (40)	maximum $r = V(4)$ to be considered in this calculation
	W(41)	A1 = vector in INTM routine
I		If $W(41) = 0$ routine will use predictor corrector with variable $V^{r}(24)$
•		If W(41) = 2 will use Runge-Kutta with fixed W(24)
I		If W(41) = 2 will use predictor-corrector with fixed W(24)
l		If $W(41) = 1$ or 2 then $W(42)$ through $W(47)$ are ignored but must have some value.
=		If W(41) = 0 they are not ignored
í I	W (42)	A2 = E upper bound on truncation error. See upper bound Equation (10) Appendix A in the INT and INTM subroutine
	W(43)	A3 = M is value from which lower bound E is calculated LBE = UBE/M in subroutine INT
1	W (44)	A4 = A as used in truncation error test EQ (10) in subroutine INT
•	W(45)	A5 = upper bound on mesh size (If = 0 no upper bound as long as within error range)
•	W (46)	A6 = lower bound on mesh size (If = 0 lower bound = 0)
ļ	W (47)	A7 = β , that is, 0 is less than β less than 1. It is used to decrease or increase mesh size by dividing or multiplying current integration mesh being used
1	W (48)	smallest attenuation to be considered
	W(49)	initial refraction index = W(1)
	W (50)	initial absorption kappa x = W(2)

Table 2. Nomenclature Describing the V and W Vectors. (Page 4 of 6)

W(51)	initial attenuation = A
W(52)	x_R (km) (Radar coordinate in geomagnetic coordinate system)
W(53)	y_R (km) (Radar coordinate in geomagnetic coordinate system
W(54)	z_R (km) (Radar coordinate in geomagnetic coordinate system
W(55)	$R = \sqrt{(x_R - x)^2 + (y_R - y)^2 + (z_R - z)^2 (km)}$
W (56)	$\Delta R = c\tau - W(55) = (2.99791 \times 10^5)[V(11)] - W(55) \text{ km}$
W(57)	new elevation angle E in degrees
W (58)	$\Delta E = W(57) - W(17)$ degrees
W (59)	$2[(W(1))(W(2))]/W(1)^{2}-W(2)^{2}$
W(60)	slant range at r, θ, ϕ
W(61)	angle A at r, θ,Φ
W(62)	elevation angle E at r, θ,φ
W (63)	assigned value to k; if 1 then control is on radius; if 2 then control is on range W(28); if 3 then control is on slant range W(60)
W(64)	value of Z
W(65)	value of Y
W (66)	value of X
W(67)	location of sign which determines the calculation for ordinary or extraordinary ray
W (68)	value of V(4) above which RINDEX is to print R vector

Table 2. Nomenclature Describing the V and W Vectors. (Page 5 of 6)

W(69)	value W(1) below which R vector is printed if W(68) = 0
W(70)	number of performed integrations
W(71)	a in COLFRZ 100-200 km
W (72)	b in COLFRZ 100-200 km
W(73)	a in COLFRZ 200-300 km
W(74)	b in COLFRZ 200-300 km
₩(75)	a in COLFRZ 300-400 km
W (76)	b in COLFRZ 300-400 km
W(77) to W(25	50) unassigned in this program

Table 2. Nomenclature Describing the V and W Vectors.
(Page 6 of 6)

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SWITCH I DOWN CALCULATE EXTRA-ORDINARY RAY ONLY
SENSE SWITCH 2 DOWN CALCULATE EXTRA-ORDINARY RAY
SENSE SWITCH 3 DOWN WILL EXIT AFTER COMPLETING THIS RAY
SENSE SWITCH 4 WITH SENSE SWITCH 6 DOWN WILL EXIT
SENSE SWITCH 4 WITH SENSE SWITCH 6 DOWN WILL EXIT
FROGESS INT, V(3), V(2), V(4)-RO, V(5), V(5), V(6), V(7), V(8), V(9), W(1), W(2), W(6), W(2), W(6), V(7), V(8), V(9), W(1), W(1), W(1), W(1), V(1), V(2), V(4)-RO, V(5), V(6), V(6), V(1), W(1), W(1), W(1), V(1), V(2), V(4)-RO, V(1), V(2), V(4), V
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B. FUNCTION SLANTR

This function is used to calculate the non-deviated ray path between the transmitter, R, and the starting point, S, (See Figures 1 and 2) from the given input data describing the problem under consideration. The input data should be designed in a manner that this assumption is true.

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C. FUNCTION QATAN

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D. FUNCTION ARCOS

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E. SUBROUTINE COORD

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F. SUBROUTINE DAUX

Subroutine DAUX is used to define the differential equations that are to be numerically integrated. As a result the previously described ray trace equations are defined in this subroutine.

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G. SUBROUTINES INT AND INTM

This is a generally available SHARE program which permits the numerical integration of a chosen set of first order non-linear differential equations. It can be operated in three possible numerical integration modes (a) Runge-Kutta with a fixed integration mesh size, (b) fourth order Adams-Moulton with a fixed integration mesh size, and (c) fourth order Adams-Moulton with a variable integration mesh size that is controlled by an error sensing routine. Because no FORTRAN source program listing is available, a SHARE description of this routine is given along with a SAP listing.

IDENTIFICATION

RW INT, Adams-Moulton, Runge-Kutta Integration 704 - FORTRAN SAP Language Subroutine Space Technology Laboratories, Robert Causey and Werner L. Frank, November 30, 1958

ABSTRACT

FORTRAN version of RW-DE2F which integrates a system of N simultaneous, first order, ordinary differential equations. Option of using either 4th order Runge-Kutta method or 4th order predictor-corrector method (Adams-Moulton) is provided. Also option of automatic error control with variable step-size is provided. Input and output are single precision but double precision is used internally to control round-off errors. Requires 12N + 3 cells for data and 693 words for program.

PURPOSE

This FORTRAN subprogram integrates a set of N simultaneous, first order differential equations. It is the FORTRAN version of the standard subroutine RW-DE2F.

RESTRICTIONS

This program has two distinct entries, one for set up and the second for performing the integration steps. The user must supply a FORTRAN subprogram (with the name DAUX) which evaluates the derivatives y^{\prime} .

METHOD

The user has the option of using either a fourth order Runge-Kutta method or the fourth order Adams-Moulton method with a fixed stepsize. There is also a variable step-size mode.

While input and output to this routine are single precision, double precision is used internally to control round-off errors. Truncation error is controlled either by choosing an appropriate step-size, or by using the variable step-size mode of operation.

For details of the method see RW-DE2F.

USAGE

a. Calling Sequence for set up (performed prior to initiating the integration).

CALL INT (V, N, A1, A2, A3, A4, A5, A6, A7)

Where V is a region of at least dimension 12N + 3

N is the number of equations

Al is the option word

A2 is E

A3 is M

A4 is A

A5 is hmax

A6 is hmin

A7 is B

For meaning of A1 - A7 see Appendix A and B of RW-DE2F.

Region V contains the following information prior to Set Up entry.

V(2) = x, initial value of independent variable

V(3) = h, value of step-size

$$\begin{array}{c} V(4) = y_1 \\ \vdots \\ V(3+N) = y_N \end{array}$$
 values of dependent variables y_1

$$\begin{cases}
V(4+N) = y_1' \\
\vdots \\
V(3+2N) = y_N'
\end{cases}$$
 values of the derivatives y_i to be supplied by the auxiliary DAUX.

Note: This region and the parameter N should be placed in COMMON since it is necessarily referred to in the main program and in the auxiliary. The cell V (1) is set up by the subprogram RW INT and will contain N scaled at 35.

b. Calling Sequence for integrating one step.

CALL INTM

No arguments are required for this statement.

SPACE REQUIRED

693 cells

CHECKOUT

This routine has been extensively tested on several check problems. In all cases the errors were approximately equal to their expected values, and there were no indications that round-off errors accumulate rapidly.

METHOD

References:

1. S. D. Conte and J. Titus, An interpretive floating point sub-routine for the solution of systems of ordinary differential equations, Appendix I, Proc. Math. Committee of Univac Scientific Exchange Meeting, Nov. 21-22, 1957 (Obtainable from Remington Rand Univac, St. Paul, Minnesota).

2. E. K. Blum, A modification of the Runge-Kutta fourth-order method, Appendix H, Proc. Math. Committee of Univac Scientific Exchange Meeting, Nov. 21-22, 1957.

In this routine the user is allowed an option of using either the Runge-Kutta classical fourth-order method as modified by E. K. Blum [Ref. (2)] or the Adams-Moulton predictor-corrector method using the Runge-Kutta method for starting the process. Let the system of equations to be solved be given in the form

(1)
$$\begin{cases} y_i' = f_i(x, y_1, y_2, ..., y_N) \\ y_i(x_0) = y_{i0} \end{cases}$$
 $i = 1, 2, ..., N.$

Let y_{in} be the value of y_i at $x = x_n$ and f_{in} the derivation of y_i at $x = x_n$ and let h be the increment (step-size) of the independent variable x. The classical Runge-Kutta fourth-order method uses the formulas

$$k_{i1} = h f_i \left(x_n, y_{in} \right) ,$$

$$k_{i2} = h f_i \left(x_n + \frac{1}{2} h , y_{in} + \frac{1}{2} k_{i1} \right) ,$$

$$k_{i3} = h f_i \left(x_n + \frac{1}{2} h , y_{in} + \frac{1}{2} k_{i2} \right) ,$$

(2)
$$k_{i4} = h f_{i} \left(k_{n} + h, y_{in} + k_{i3} \right),$$
$$y_{i, n+1} = y_{n} + \frac{1}{6} \left(k_{i1} + 2k_{i2} + 2k_{i3} + k_{i4} \right),$$

The following formulas (we omit the subscript i for notational simplicity) were derived by E. K. Blum to control the growth of round-off errors.

off errors.

$$\begin{cases}
z_{o} = y_{n}, \\
q_{o} = q_{4n}
\end{cases}$$

$$\begin{cases}
P_{o} = h f(x_{n}, z_{o}) \\
r_{1} = L^{(1)}R^{(1)} \left[\frac{1}{2}P_{o} - q_{o}\right], \\
z_{1} = z_{o} + r_{1}, \\
q_{1} = 3r_{1} - \left[\frac{1}{2}P_{o} - q_{o}\right], \\
\end{cases}$$

$$\begin{cases}
P_{1} = h f(x_{n} + \frac{1}{2}h, z_{1}), \\
r_{2} = L^{(2)}R^{(2)} \left[\frac{1}{2}P_{1} - \frac{1}{2}q_{1}\right], \\
z_{2} = z_{1} + r_{2}, \\
q_{2} = -r_{2} - \frac{1}{3}q_{1} + \frac{1}{2}P_{1}, \\
\end{cases}$$

$$\begin{cases}
P_{2} = h f(x_{n} + \frac{1}{2}h, z_{2}), \\
r_{3} = L^{(3)}R^{(3)} \left[P_{2}\right], \\
z_{3} = z_{2} + r_{3}, \\
q_{3} = -r_{3} + q_{2},
\end{cases}$$
(6)

(7)
$$\begin{cases} P_3 &= h f(x_n + h, z_3) + 2 P_2, \\ r_4 &= L^{(4)} R^{(4)} \left[\frac{1}{6} P_3 + q_3 \right], \\ y_{n+1} &= z_4 = z_3 + r_4, \\ q_{4, n+1} &= 3 \left[r_4 - \left(\frac{1}{6} P_3 + q_3 \right) \right], \end{cases}$$

where R^(m), L^(m) denote operators which shift right m places or left m places respectively and q₄₀ is taken to be zero to start the computation. (See Ref (2) for a complete description of this method.) Formulas (3) - (7) are those used in this routine.

The Adams-Moulton predictor-corrector formulas for the system (1) are

(8)
$$y_{i, n+1}^{(p)} = y_{in} + \frac{h}{24} \left[55 f_{in} - 59 f_{i, n-1} + 37 f_{i, n-2} - 9 f_{i, n-3} \right]$$

(9)
$$y_{i, n+1}^{(c)} = y_{in} + \frac{h}{24} \left(9 f_{i, n+1}^{(p)} + 19 f_{in} - 5 f_{i, n-1} + f_{i, n-2} \right)$$
.

The corrector formular (9) is applied only once so that only two derivative evaluations are needed for each Adams-Moulton integration step. The starting values needed in (8) are obtained using the Runge-Kutta-Blum (RKB) method.

The Adams-Moulton method may be used either with a fixed step-size or with a variable step-size. The step-size to be used in the variable mode is determined as follows. Let

$$E_{n+1} = \max_{i} \frac{\begin{vmatrix} y_{i,n+1}^{(p)} - y_{i,n+1}^{(c)} \\ \frac{14D_{i}}{1} \end{vmatrix}}{14D_{i}},$$

$$D_{i} = \max_{i} \left\{ \begin{vmatrix} y_{i,n+1}^{(c)} \\ y_{i,n+1}^{(c)} \end{vmatrix}, A \right\},$$

where A>0. The user will specify an upper bound \overline{E} on the truncation error estimate E_{n+1} . This is equivalent to specifying the number of significant figures which the user desires to preserve locally throughout the integration. There must also be specified a constant M>0

from which a lower bound $E = M^{-1} \overline{E}$ is obtained. M should normally range from 50 to 150. The interval will then be decreased, left as it is, or increased according as the following inequalities hold:

(11a) If
$$E_{n+1} \ge \overline{E}$$
, the interval is reduced to βh (0 < β <1

(11b) If
$$\underline{E} \leqslant \underline{E}_{n+1} < \overline{\underline{E}}$$
, the interval size is kept fixed.

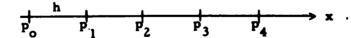
(11c) If
$$E_{n+1} < E$$
 for 3 successive steps, the step-size is increased to $\frac{1}{\beta}$ h.

Normally, the routine will take $\beta = 1/2$, unless β is otherwise specified. The constant A in (10) is used to prevent unnecessary reductions in |h| whenever $|y_{i,n+1}|$ is small. Normally the routine

will set A=1. However, some other value for A may be specified by the user if he desires to use some other characteristic length for A. While the test based on (10) will guarantee that the local error does not exceed \overline{E} , the cumulative error will usually exceed \overline{E} . Hence, \overline{E} should be chosen small enough to allow for an accumulation of truncation error. Normally \overline{E} should be in the range $10^{-8} \leqslant \overline{E} \leqslant 10^{-3}$.

After an interval is increased, the program prevents increasing again until 6 more points have been completed. However, the program may decrease the interval as often as necessary.

Starting values for the Adams-Moulton formulas are always obtained using the RKB method whenever the interval size is changed, just as at the beginning of an integration. Consider the following diagram of the axis of the independent variable x



If the values of the y_i are computed at the points p_1 , p_2 , and p_3 using the RKB method and the truncation error test (11) calls for decreasing |h| at the point p_4 , then the routine returns to the point p_0 and again computes three new points with the RKB method using the decreased value of |h|. If on the other hand (11a) holds at p_4 and the y_i at p_3 had been computed using the AM method, then the routine returns to the point p_3 for a new start. If the inequality in

(11c) is not satisfied at p_1 , but is satisfied at p_2 , p_3 , and p_4 , then a new start is initiated at p_4 with the increased value of |h|.

The user must provide a starting value for h and he may, if desired, specify a maximum value of |h| beyond which the routine will not increase |h| and a minimum value of |h| below which it will not decrease |h|. Negative values of h may be supplied for backward integration.

Both the RKB method and the AM method incorporate round-off control features. This is performed in the RKB method by carrying the q's in formula (3)-(7). In the AM method this is done by keeping the y_{in} in double precision forming the sums $y_{in} + \nabla y_{in}$ in (8) and (9) in double precision. The derivative calculations are all performed in single precision. Both procedures have shown to be very effective in controlling the growth of round-off errors.

USAGE AND CODING INFORMATION (APPENDIX B)

There are two entries to this routine. The first must be used once at the beginning to set up the routine for integration of a given set o N differential equations. The second entry may be used any number of times after the first to integrate all y_i from x to x + h. The first entry has the following calling sequence.

Loc.	Instruction	Comment
A1-2	TSX DE2F, 4	Setup entry
A1-1	PZE T, O, V	Parameter word with addresses
Al	(Binary integer)	Option word (= 0 or 1 or 2)
A2	(Floating-point number)	E Truncation error testing
A3	(Floating-point number)	M information
A4	(Floating-point number)	A
A 5	(Floating-point number)	h]
A 6	(Floating-point number)	h Bounds on h, if any
A7	(Floating-point number)	β - Increase or decrease h factor
A7 +1	Return	

The eight parameter words have the following meaning, (A1-1): V is the address of the first word of a block of 12N + 3 cells, reserved by the user, with the arrangement

Loc.	Contents	
v	N	Fixed point binary integer, point at right
V + 1	x	Value of independent variable in floating point
V + 2	h	Value of step size in floating point
V + 3	Υı]	•
V + 4	y ₂	
•	• 1	
•	. }	Values of the y _i in floating point
•		
V + N + 2	yΝ	

Loc.	Contents
V + N + 3	y_1'
V + N + 4	y 2
•	}
•	1
V + 2N + 2	y _N
V + 2N + 3	-
etc.	}

Locations where the user's auxiliary subroutine must place the derivatives y_1^{i} .

10 N cells of temporary storage

Note: If the Runge-Kutta only option (see under Al below) is used, it is only necessary to reserve a block of 4N + 3 cells.

Before executing the setup entry, the user must have already placed the appropriate numbers in cells V through V + N + 2.

The address V in the entry point of an auxiliary subroutine which the user must provide to evaluate the derivatives y_i and store them in cells V + N + 3 through V + 2N + 2 as shown above. This auxiliary subroutine is entered by the calling sequence

Loc.	Instruction
A1-2	TSX V, 4
A1-1	Return

The setup entry uses the auxiliary subroutine to evaluate the derivatives for the initial data.

(A1): The option word may have any one of three values which designate three different modes of operation for RWINT

Al = 0 designates the predictor-corrector variable h mode

Al = 1 designates the fixed h Runge-Kutta only mode

A1 = 2 designates the fixed h predictor-corrector mode

For A1 = 1 or 2, the contents of A2 through A7 may be arbitrary.

(A2): This cell contains the upper bound E>0 for the truncation error testing done in the predictor-corrector variable h mode. $(10^{-8} < E < 10^{-3})$

(A3): This cell contains the number M>0 from which the lower bound E is calculated. If A3 = 0, M is set equal to 100.

(A4): This cell contains the number A>0 used to designate a fixed-point truncation error test as described in Appendix A. If A4=0, A is set equal to 1.

(A5): This cell may contain the upper bound $h_{max} > 0$ for |h|. If A5 = 0, this means that there is to be no upper bound for |h|.

(A6): This cell may contain the lower bound $h_{min} > 0$ for |h|. If A6 = 0, this means that there is to be no lower bound for |h|.

(A7): This cell may contain the factor $1 > \beta > 0$ used to increase or decrease |h|. If A7 = 0, β is set equal to 1/2.

The integration entry is quite simple and has the calling sequence

Loc.	Instruction
A1-2	TSX DE2F + 1, 4 Integration entry
A1-1	Return

Ordinarily, after execution of the integration entry, all y_i assume new values, x will have been advanced to the value x + h and h will be unchanged. However, in the variable h mode, three other things can happen. (1) if the truncation error test indicates that h should be increased, h will have been changed to $\beta^{-1}h$ unless $\beta^{-1}h > h_{max}$. If the truncation error test indicates that h should be decreased, then h will have been changed to $\beta^{-1}h$ unless $\beta h < h_{min}$ and either (2) y_i and x will remain as they were before entry or (3) x will be changed to x - 3h and the corresponding y_i values will occupy calls V + 3 through V + N + 2. Case (3) can only happen when successive decreases in h are called for. On exit the values y_i in V + N + 3 etc. are always those which correspond to the x and y_i in V + 1 and V + 3 etc.

The integration entry must be used for each integration step. In the variable h mode, a particular integration step may involve either AM or RKB integration but not both. In the fixed h predictor-corrector mode, the first three integration entries involve RKB integration and all subsequent ones involve AM integration.

Whenever an integration step involves AM integration, the truncation error estimate E_{n+1} is in the accumulator on exit. Zero is always placed in the accumulator if the step involved RKB integration.

The setup entry may be used again at any time to set up another problem or to change the mode of operation.

In addition to the auxiliary subroutine for derivative evaluation and the 12N + 3 cells for data storage, the storage requirements are 693 words for RWINT plus 2 words of COMMON.

```
INT 0001
      ORG 0
                                                                            INT 0002
DAUX BCD 1DAUX
                                                                            NT 0003
      HTR
                                                                            INT 0004
      HTR
                                                                            INT 0005
      HTR
                                                                            INT 0006
INT
      SXD INT-3,4
                                                                            INT 0007
      SXD INT-2.2
                                                                            INT 0008
      5XD INT-1+1
                                                                            INT 0009
      CLA 1,4
                                                                            INT 0010
      STA REV1+1
                                                                            INT 0011
      STA REV1+2
                                                                             INT 0012
      STA A2
                                                                             INT 0013
      CLA 2+4
                                                                             INT 0014
      STA A1
                                                                             INT 0015
A1
      CLA
                                                                             INT 0016
      ARS 18
                                                                             INT 0017
AZ
      STO
                                                                             INT 0018
      ALS 2
                                                                             INT 0019
      STO C
                                                                             INT 0020
      ALS 1
                                                                             INT 0021
                             -12N
      ADD C
                                                                             INT 0022
      ADD C1
                              12N+2
                                                                             INT 0023
      STO C
                                                                             INT 0024
      CLA 1.4
                                                                             INT 0025
      SUB C
                                                                             INT 0026
      STA PARI
                                                                             INT 0027
      CLA C
                                                                             INT 0028
      ARS 1
                                                                             INT 0029
       STO C
                                                                             INT 0030
       ADD PARL
                                                                             INT 0031
       STA REVI
                                                                             INT 0032
       STA REV1+3
                                                                             INT 0033
      LXA CZ+1
                                                                             INT 0034
      CLA 3.4
A4
                                                                             INT 0035
       STA A5
                                                                             INT 0036
A5
       CLA
                                                                             INT 0037
       STO PAR8+1+1
                                                                             INT 0038
       TX1 A6.4.-1
                                                                             INT 0039
       TIX A4.1.1
A6
                                                                             INT 0040
       CLA PARZ
                                                                             INT 0041
INT 0042
      LRS 18
STO PAR2
                                                                             INT 0043
       CLA DAUX
                                                                             INT 0044
       STA AUX+2
                                                                             INT 0045
       TSX REV.4
                                                                             INT 0046
       TSX DE2F+4
                                                                             INT 0047
     PZE 0.0.AUX
PAR1
                                                                             INT 0048
PAR2 PZE
                                                                             INT 0049
     PZE
PAR3
                                                                             INT 0050
     PZE
PAR4
                                                                             INT 0051
PAR5
      PZE
                                                                             INT 0052
      PZE
PAR6
                                                                             INT 0053
PART PZE
```

```
PARS PZE
                                                                           1NT 0054
       TSX REV.4
                                                                           INT 6055
       LXD INT-2.2
                                                                           INT 0056
       LXD INT-1.1.
                                                                           INT 0057
       LXD 1NT-3.4
                                                                           INT 0058
       TRA 10.4
                                                                           INT 0059
 INTM SXD INT-3:4
                                                                           INT 0060
       SXD INT-2.2
                                                                           INT 0061
       SXD INT-1+1
                                                                           INT 0062
       TSX REV+4
                                                                           INT 0063
       TSX DE2F+1.4
                                                                           INT 0064
       TSX REV.4
                                                                           INT 0065
       LXD INT-3.4
                                                                           INT 0066
       LXD INT-2.2
                                                                           INT 0067
       LXD INT-1.1
                                                                           1NT 0068
       TRA 1.4
                                                                           INT 0069
REV
       LXA C+1
                                                                           INT 0070
       LXD C1.2
                                                                           INT 0071
 REV1
       CLA 0.1
                                                                           INT 0072
       LDQ 0.2
                                                                           INT 0073
       STO 0.2
                                                                           INT 0074
       STO 0.1 -
                                                                           INT 0075
       TX1 REV3.2.1
                                                                           INT 0076
REV3
       TIX REV1.1.1
                                                                           INT 0077
       TRA 1.4
                                                                           INT 0078
AUX
       SXD C3.4
                                                                           INT 0079
                                                                           INT 0080
       TSX REV.4
       T5X 0.4
                                                                           INT 0081
       TSX REV.4
                                                                           INT 0082
       LXD C3.4
                                                                           INT 0083
       TRA 1,4
                                                                           INT 0084
                                                                           INT 0085
       PZE
C
                                                                           INT 0086
C1
       DEC 2
                                                                           INT 0087
       DEC 7
 C2
                                                                           INT 0088
       REM FLOATING POINT ADAMS-MOULTON. RUNGE-KUTTA INTEGRATION
                                                                           INT 0089
                              SETUP ENTRY
DE2F
       TRA DE2F+0293
                                                                           INT 0090
       SXD DE2F+0240+1
                                    INTEGRATION ENTRY
                                                                           INT 0091
       SXD DE2F+0241+2
                                                                           INT 0092
       SXD DE2F+0242+4
                                                                           INT 0093
                              SWITCH 1
       TRA DE2F+0005
                                                                            INT 0094
       CLA DE2F+0285
                              3 TO ACC
                                                                            INT 0095
       CAS DE2F+0228
                              TEST ALPHA
                                                                            INT 0096
       TRA DE2F+0186
                                                                            INT 0097
       TRA DE2F+0175
                                                                            INT 0098
       LXA DE2F+0229+1
                                 Y PRIMES TO D
                                                                            INT 0099
                                                                            INT 0100
       CLA #+1
                                                                            INT 0101
       STO #.1
                                  AND
                                                                            INT 0102
       CLA *+1
                                 DOUBLE PRECISION
                                                                           INT 0103
       STO *.1
                                                                           INT 0104
       CLA *.1
                                  YS TO TS2
                                                                            INT 0105
       STO *.1
                                                                           INT 0106
                                     END 3F LOOP
       TIX DE2F+0010+1+1
```

LXA DEZF	+0229•1			INT		
LDQ DE2F	+0223		•	INT		
FMP *+1		D3 SUB I		INT		
STO COMM	ON+000			INT	0110	•
CLA *+1		PLANT Y SUB	1	INT		-
STO DEZF	+0267 IN	SFA .		INT	0112	!
CLA *+1		SUBROUTINE		INT	0113	,
STO DE2F	+0268			INT	0114	
LDQ DE2F				INT	0115	•
FMP *,1		D2 SUB 1		INT	0116	5
FAD COMM	ON+000 ·			INT	0117	•
STO COMM				INT	0118	3
LDQ DE2F				INT	0119)
FMP *•1		D1 SUB I		INT	0120)
FAD COMM	ION+000			INT	0121	L
STO COMM				INT	0122	2
LDQ DE2F				INT	0123	3
FMP +.1	. 0220	D SUB I		INT	0124	+
FAD COMM	40N+000			INT	0129	5
STO COMM				INT	0126	5
LDQ *	LO	AD. H		INT	0127	7
FMP COMM	T1	LTA YI UPPER P		INT	012	3
	+0269+2	ADD TO YI		INT	0129	9
\$70 #.1	+020312	STORE IN TS1		INT	0130	0
• • • • • • • • • • • • • • • • • • • •	+0018+1+1	END OF LOOP		INT	013	1
	400184141			INT	013	2
CLA * FAD *	X4	М		INT	013	3
	~ ~	•		INT	013	4
STO #	F+0220.1			INT	013	5
• • • • • • • • • • • • • • • • • • • •	F+0229•1		•	INT	013	6
CLA *+1				INT	013	7
STO *,1		END OF LOOF		INT	013	8
, , , , , , , , , , , , , , , , , , , ,	F+0046+1+1	ALUATE DERIVATIV		INT		
TSX 0.4		ACONIC DENIUNITI		INT	014	0
	F+0229+1			•	014	-
LDQ DE2	+40221	D2 SUB I			014	_
FMP +,1		02 300 1		INT	014	3
STO COM	MON+OOO	YI FROM TS2			014	
CLA *+1	T. 0047	11 PROF 102			014	
STO DE2		DP EXT.	•	INT		
CLA ++1	_	OF EATS			014	
STO DE2					014	
LDQ DE2		D1 SUB1			014	
FMP +.1		01 3091			015	
FAD COM					015	_
STO COM				•	015	
LDQ DE2		D #1100			015	
FMP +.1		D SUBI			019	
	MON+000				01:	
	MON+000				01:	
TDO DES		-			01:	
FMP +,1		Y PRIME SUBI		•	01	
,	MON+000			• • • •	01	
STO COM	MON+000			144	AT:	, 7

		1		
LDQ	#	LOAD H	INT	0160
FMP	COMMON+000	DELTA YI UPPER C		0151
TSX	DE2F+0269+2	ADD TO Y I	INT	0162
STO	*,1			0163
STO	*.1			0164
	DE2F+0051+1+1	END OF LOOP	_	0165
	DE2F+0230			0166
	DE2F+0229+1			0167
CLA				0168
	DE2F+0243			0169
CLA			-	0170
	DE2F+0244		-	0171
-	DE2F+0246+2		-	0172
		END OF LOOP		0172
	DE2F+0078+1+1 DE2F+0085	SWITCH 2		0174
		SWITCH 2		
	DE2F+0234			0175
	DE2F+0230		-	0176
• .	DE2F+0090			0177
•	DE2F+0089	ACADE 4 C	_	0178
	DE2F+0115	DECREASE H SWITCH		0179
	DE2F+0235		-	0180
	DE2F+0230			0181
		INCREASE H SWITCH		0182
	DE2F+0211			0183
	DE2F+0211		-	0184
_	DE2F+0229 • 1			0185
	*,1			0186
	*•1	D2 TO D3		0187
	*•1	·		0188
	**1	D1 T0 D2		0189
	* •1			0190
STO	* • 1	D TO D1		0191
TIX	DE2F+0096 • 1 • 1	END OF LOOP	_	0192
CLA	DE2F+0228		_	0193
ADD	DE2F+0284	ALPHA PLUS ONE		0194
STO	DE2F+0228		-	0195
TSX	0.4	EVALUATE DERIVATIVES		0196
CLA	DE2F+0230			0197
FDP	DE2F+0290	E	INT	0198
STO	COMMON+000		INT	0199
CLA	COMMON+000	GET E IN ACC	INT	0200
LXD	DE2F+0240+1		INT	0201
LXD	DE2F+0241+2		INT	0202
LXD	DE2F+0242+4		INT	0203
TRA	1,4	EXIT	INT	0204
CLA	•		INT	0205
SSP			INT	0206
	DE2F+0238	TEST H WITH HMIN	INT	0207
	DE2F+0121		INT	0208
-	DE2F+0121		INT	0209
	DE2F+0211		INT	0210
LDQ			INT	0211
_		STORE OLD H	INT	0212
	- · -	_		

```
1NT 0213
         FMP DE2F+0239
                                BETA TIMES H
                                                                             INT 0214
         STO *
                                                                             INT 0215
         CLA DE2F+0285
                                TEST ALPHA
                                                                             INT 0216
         CAS DE2F+0228
                                                                             INT 0217
         HTR DE2F+0127
                                                                             INT 0218
         TRA DE2F+0130
                                                                              INT 0219
         TRA DE2F+0144
         LDQ DE2F+0289
                                                                             INT 0220
         FMP DE2F+0233
                                4H
                                                                              INT 0221
                                                                             INT 0222
         CHS
                                X-4H
                                                                              INT 0223
         FAD *
                                                                             INT 0224
         STO *
         LXA DE2F+0229+1
                                                                             INT 0225
                                                                             INT 0226
         CLA * 1
١
                                                                             INT 0227
         STO #+1
                                                                             INT 0228
         CLA * + 1
                                                                             INT 0229
         STO * 1
                                                                             INT 0230
         CLA * 1
                                                                              INT 0231
         sto *.1
                                    END OF LOOP
                                                                              INT 0232
         TIX DE2F+0136+1+1
                                JUMP TO SET ALPHA
                                                                              INT 0233
         TRA DE2F+0161
                                                                              INT 0234
         CLA DE2F+0233
                                                                              INT 0235
         CHS
                                                                              INT 0236
         FAD #
                                X-H
                                                                              INT 0237
         STO #
                                                                              INT 0238
         LXA DE2F+0229+1
                                                                              INT 0239
         CLA * . 1
                                                                              INT 0240
                                    D TO YI PRIME
         STO *.1
                                                                              INT 0241
                                    TS2 TO YI
         CLA *.1
                                                                              INT 0242
         STO *.1
                                                                              INT 0243
         CLA *+1
                                                                              INT 0244
         510 *.1
                                                                              INT 0245
         TIX DE2F+0149+1+1
                                                                              INT 0246
         LXA DE2F+0229+1
                                                                              INT 0247
                                CONVERT DP YI
         LDQ DE2F+0291
                                                                              INT 0248
                                    TO YI. QI
         FMP # . 1
                                                                              INT 0249
         STO * 1
                                     END OF LOOP
                                                                              INT 0250
         TIX DE2F+0157+1+1
                                                                              INT 0251
         STZ DE2F+0228
                                                                              INT 0252
         STZ DE2F+0231
                                                                              INT 0253
         TRA DE2F+0107
                                                                              INT 0254
         CLA *
                                                                              INT 0255
         SSP
                                TEST H WITH HMAX
                                                                              INT 0256
         CAS DE2F+0237
                                                                              INT 0257
                                GO TO SHIFT DS
         TRA DE2F+0095
                                                                              INT 0258
         TRA DE2F+0213
                                                                              INT 0259
         TRA DE2F+0213
                                                                              INT 0260
         CLA #
                                H DIVIDED BY BETA
                                                                              INT 0261
         FDP DE2F+0239
                                                                              INT 0262
         STO #
                                EVALUATE DERIVATIVES
                                                                              INT 0263
         TSX 0.4
         TRA DE2F+0156
                                                                              INT 0264
                                                                              tht 0265
         LXA DE2F+0229,1
```

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```
CLA *,1
                                                                     INT 0266
                                                                      INT 0267
FDP DE2F+0291
                      QI DIVIDED BY -3
                                                                     INT 0268
CLA *,1
                                                                     INT 0269
STQ COMMON+000
                                                                     INT 0270
FAD COMMON+000
                                                                     INT 0271
STO *.1
STQ * , 1
                                                                     INT 0272
TIX DE2F+0176 . 1 . 1
                                                                     INT 0273
TRA DE2F+0009
                                                                     INT 0274
                           DUMMY ORDER
                                                                     INT 0275
STO * 1
                                                                     INT 0276
LDQ DE2F+0229
                                                                     INT 0277
MPY DE2F+0228
                                                                      INT 0278
STQ COMMON+000
CLA COMMON+000
                                                                      INT 0279
ADD DE2F+0185
                                                                      INT 0280
                                                                      INT 0281
STO DE2F+0194
LXA DE2F+0229+1
                                                                      INT 0282
                                                                      INT 0283
CLA *,1
                       STORE DERIVATIVES
                                                                      INT 0284
STO 0
TIX DE2F+0193 • 1 • 1
                                                                      INT 0285
CLA DE2F+0228
                                                                      INT 0286
                                                                      INT 0287
TNZ DE2F+0204
                                                                      INT 0288
LXA DE2F+0229.1
                                                                      INT 0289
CLA *+1
                                                                      INT 0290
STO * + 1
CLA * . 1
                                                                      INT 0291
STO * .1
                                                                      INT 0292
TIX DE2F+0199+1+1
                                                                      INT 0293
CLA DE2F+0228
                                                                      INT 0294
                                                                      INT 0295
                       ALPHA PLUS ONE
ADD DE2F+0284
                                                                      INT 0296
STO DE2F+0228
                                                                      INT 0297
                         RUNGA-KUTTA ENTRY
TSX DE2F+0454,4
                                                                      INT 0298
STZ COMMON+000
                                                                      INT 0299
                       ZERO TO ACC
CLA COMMON+000
                                                                      INT 0300
TRA DE2F+0111
                                                                      INT 0301
STZ DE2F+0231
                                                                      INT 0302
TRA DE2F+0095
                                                                      INT 0303
CLA DE2F+0231
                                                                      INT 0304
                       R+1
ADD DE2F+0284
                                                                      INT 0305
STO DE2F+0231
                                                                      INT 0306
CAS DE2F+0292
                                                                      INT 0307
HTR DE2F+0217
                                                                      INT 0308
TRA DE2F+0170
                                                                      INT 0309
TRA DE2F+0095
                                                                      INT 0310
DEC 2.291666667
                                                                      INT 0311
DEC -2.458333333
                                                                      INT 0312
DEC 1.541666667
                                                                      INT 0313
DEC -3.75E-1
                                                                      INT 0314
DEC 3.75E-1
                                                                      INT 0315
DEC 7.916666667E-1
                                                                      INT 0316
DEC -2.083333333E-1
                                                                      INT 0317
DEC 4.166666667E-2
                                                                      INT 0318
BSS 4
```

```
BSS 8
                                                                     INT 0319
                                                                     INT 0320
855 3
                       HI UPPER C
                                                                     INT 0321
HTR
                       YI UPPER P
                                                                    INT 0322
HTR
HTR
                       DIVISOR
                                                                    INT 0323
CLA DE2F+0243
                       ENTRY
                                                                     INT 0324
SSP
                                                                    INT 0325
CAS DE2F+0236
                       TEST FOR DIVISOR
                                                                    INT 0326
TRA DE2F+0254
                                                                     INT 0327
TRA DE2F+0254
                                                                     INT 0328
CLA DEZF+0236
                                                                     INT 0329
STO DE2F+0245
                                                                     INT 0330
TRA DE2F+0255
                                                                     INT 0331
STO DE2F+0245
                                                                    INT 0332
                                                                    INT 0333
CLA DE2F+0244
FSB DE2F+0243
                                                                    INT 0334
SSP
                                                                    INT 0335
FDP DE2F+0245
                                                                    INT 0336
STQ COMMON+000
                                                                    INT 0337
CLA DE2F+0230
                                                                    INT 0338
CAS COMMON+000
                                                                    INT 0339
TRA 1.2
                                                                    INT 0340
                                                                   · INT 0341
TRA 1.2
                                                                    INT 0342
CLA COMMON+000
                                                                    INT 0343
STO DE2F+0230
TRA 1.2
                                                                    INT 0344
                                                                    INT 0345
HTR
                       A1
                                                                    INT 0346
HTR
                       A2
                       ENTRY.
                                                                    INT 0347
INT 0348
UFA DE2F+0267
STO DE2F+0267
                                                                     INT 0349
                       SPECIAL
STQ COMMON+000
CLA COMMON+000
                       FLOATING
                                                                     INT 0350
                                                                     INT 0351
                       ADDITION
UFA DE2F+0268
                                                                     INT 0352
FAD DE2F+0267
                       SUBROUT INE
                                                                     INT 0353
TRA 1.2
                                                                     INT 0354
TRA DE2F+0005
                       SWITCH 1. A LEG
                       SWITCH 1. B LEG
                                                                     INT 0355
TRA DE2F+0207
                                                                     INT 0356
                       SWITCH 2. A LEG
TRA DE2F+0085
                       SWITCH 2. B LEG
                                                                     INT 0357
TRA DE2F+0095
                       DECREASE H SWITCH. TEST LEG
                                                                     INT 0358
TRA DE2F+0115
                       DECREASE H SWITCH. NO TEST LEG
                                                                     INT 0359
TRA DE2F+0121
                       INCREASE H SWITCH. TEST LEG
                                                                     INT 0360
TRA DE2F+0164
                       INCREASE H SWITCH. NO TEST LEG
                                                                     INT 0361
TRA DE2F+0213
                                                                     INT 0362
DEC 1
DEC 3
                                                                     INT 0363
                                                                     INT 0364
DEC 1.
DEC 5E-1
                       ONE HALF
                                                                     INT 0365
DEC 100.
                                                                     INT 0366
                                                                     INT 0367
DEC 4.
DEC 14.
                                                                     INT 0368
                                                                     INT 0369
DEC -3.
                       SPECIAL TEST NO. FOR R
                                                                     INT 0370
DEC 3
                                                                     INT 0371
SXD DE2F+0241.2
```

CYD	DE2F+0242+4			0372
	DE2F+0276		INT	0373
	DE2F+0004	SET	INT	0374
	DE2F+0278	SWITCHES	INT	0375
	DE2F+0084	10	INT	0376
	DE2F+0280	NORMAL	INT	0377
		POSITIONS	INT	0378
	DE2F+0089	F03111(M3		0379
	DE2F+0282			0380
-	DE2F+0092		-	0381
CLA			•	0382
PAX		ARI RET ARTIAN		0383
	DE2F+0308+2	SELECT OPTION		0384
	DE2F+0346	OPTION 2		0385
	DE2F+0343	OPTION 1	•	0386
LDQ		OPTION 0		0387
	DE2F+0290	14E (UPPER)	_	0388
	DE2F+0234			0389
CLA				0390
TNZ	DE2F+0314	TEST BETA	•	0391
	DE2F+0287	•		
STO	DE2F+0239	STORE BETA		0392
CLA	4,4		•	0393
TNZ	DE2F+0318		•	0394
CLA	DE2F+0288		-	0395
STO	DE2F+0235	STORE M	•	0396
-	DE2F+0234		-	0397
	DE2F+0235			0398
	DE2F+0235	STORE 14E (LOWER)	-	0399
• -	5,4		_	0400
	DE2F+0325	TEST A	•	0401
–	DE2F+0286		•	0402
•	DE2F+0236	STORE A		0403
	6,4		-	0404
	DE2F+0331	TEST.H MAX	INT	0405
	DE2F+0283		INT	0406
	DE2F+0092		INT	0407
• • •			INT	0408
	DE2F+0335 DE2F+0237		INT	0409
• • •			INT	0410
	DE2F+0239		INT	0411
	DE2F+0237	STORE BETA (H MAX)	INT	0412
	DE2F+0237	STOKE DEIN IN MANT	INT	0413
	7,4	TEST H MIN	INT	0414
	DE2F+0340	IESI U MIN	INT	0415
	DE2F+0281		-	0416
	DE 2F+0089			0417
	DE2F+0348			0418
	DE2F+0239		-	0419
	DE2F+0238			0420
	DE2F+0348			0421
	DE2F+0277		-	0422
	DE2F+0004			0423
	DE2F+0348		_	0424
CLA	DE2F+0279		A 14 (. • •

			INT 0425
	5TO DE2F+0084		INT 0426
	CLA 1+4		INT 0427
	STO DE2F+0232		INT 0428
	STO DE2F+0352	ATTIO BY CURRALITINE	INT 0429
	TSX DE2F+0453+4	SETUP RK SUBROUTINE	INT 0430
	PZE	PAR METER WORD	INT 0431
,	STZ DE2F+0228	SET ALPHA TO ZERO	INT 0432
•	STZ DE2F+0231	R O	INT 0433
	CLA DE2F+0232		INT 0434
	STA DE2F+0357		INT 0435
i	CLA 0	CTARP N	INT 0436
•	STO DE2F+0229	STORE N	INT 0437
	CLA DE2F+0232		INT 0438
	ARS 18	A.C. 115	INT 0439
1	STA DE2F+0049	SETUP	INT 0440
	STA DE2F+0106	DERIVATIVE	INT 0441
	STA DE2F+0173	EVALUATIONS	INT 0442
	CLA DE2F+0232		INT 0443
!	ADD DE2F+0284	T=1	INT 0444
•	STA DE2F+0042		INT 0445
	STA DE2F+0044	•	INT 0446
1	STA DE2F+0133	·	INT 0447
ł	STA DE2F+0134		INT 0448
•	STA DE2F+0146		INT 0449
	STA DE2F+0147		. INT 0450
1	ADD DE2F+0284	T=2	INT 0450
	STA DE2F+0037		INT 0452
•	STA DE2F+0043		INT 0452
	STA DE2F+0070		INT 0454
1	STA DE2F+0115		1NT 0455
	STA DE2F+0121		INT 0456
_	STA DE2F+0124		INT 0457
_	STA DE2F+0164		INT 0458
	STA DE2F+0170		INT 0459
•	STA DE2F+0172	= _	INT 0460
	ADD DE2F+0284	T=3 EQUALS D	INT 0461
_	ADD DE2F+0229	D=N	INT 0462
	STA DE2F+0010		INT 0463
ı	STA DE2F+0021		INT 0464
	STA DE2F+0047		INT 0465
_	STA DE2F+0073		INT 0466
	STA DE2F+0078		INT 0467
	STA DE2F+0139		INT 0468
	STA DE2F+0152		INT 0469
•	STA DE2F+0178		INT 0470
	STA DE2F+0181		INT 0471
	STA DE2F+0199		INT 0472
	ADD DE2F+0229	D+2N	INT 0473
	STA DE2F+0012		INT 0474
	STA DE2F+0067		INT 0475
•	STA DE2F+0137		INT 0476
	STA DE2F+0150		INT 0477
6	STA DE2F+0193		THÍ ÔML
	-		

ADD DE2F+0229	D+3N	INT	0478
STA DE2F+0014		INT	0479
STA DE2F+0023		INT	0480
STA DE2F+0056			0481
STA DE2F+0074			0482
STA DE2F+0141		- ·	
			0483
STA DE2F+0154		-	0484
STA DE2F+0158		INT	0485
STA DE2F+0159		INT	0486
STA DE2F+0176		INT	0487
STA DE2F+0182		- ' ' '	0488
STA DE2F+0201			0489
ADD DE2F+0229	D+4N		0490
	DTAN	=	
STA DE2F+0040		- :	0491
STA DEZF+0046			0492
STA DE2F+0080		-	0493
ADD DE2F+0229	D+5N	INT	0494
STA DE2F+0011		INT	0495
STA DE2F+0054		INT	0496
STA DE2F+0151			0497
ADD DE2F+0229	D+6N		0498
STA DE2F+0015	D 1011		0499
STA DE2F+0153			0500
ADD DE2F+0229	D+7N	=	0501
STA DE2F+0138			0502
STA DE2F+0200		=	0503
ADD DE2F+0229	D+8N	, INT	0504
STA DE2F+0140		INT	0505
STA DE2F+0202		INT	0506
ADD DE2F+0229	D+9N	INT	0507
STA DE2F+0019	• • • • •	INT	0508
STA DE2F+0097		- `	0509
STA DE2F+0136		-	0510
			0511
STA DE2F+0185		-	
ADD DE2F+0229	D+10N		0512
STA DE2F+0026		-	0513
STA DE2F+0052		•	0514
STA DE2F+0096			0515
STA DE2F+0099		INT	0516
ADD DE2F+0229	D+11N	INT	0517
STA DE2F+0030		INT	0518
STA DE2F+0059		INT	0519
STA DE2F+0098		INT	0520
STA DE2F+0101		INT	0521
ADD DE2F+0229	D+12N		0522
STA DE2F+0013	D. 154		0523
		_	0524
STA DE2F+0034			
STA DE2F+0063		INT	-
STA DE2F+0100		•	0526
STA DE2F+0149		INT	
LXD DE2F+0241+2		INT	0528
LXD DE2F+0242+4		INT	0529
TRA 9.4	EXIT	INT	0530

TRA DE2F+0562	TO SETUP REGION	INT 0531 INT 0532
SXD DE2F+0553+1	SAVE INDEX REGISTERS	INT 0532
SXD DE2F+0554+2	FROM	INT 0534
SXD DE2F+0555+4	MAIN PROGRAM	INT 0535
CLA		INT 0536
FDP DE2F+0520	CALCULATE	INT 0537
STQ DE2F+0525	H DIVIDED BY 2	1NT 0538
STQ DE2F+0541		INT 0539
LXA DE2F+0551+2	SET PARAMETER INDEX	INT 0540
LXA DE2F+0556+1	SET N INDEX.	INT 0541
LDQ 1		INT 0542
FMP DE2F+0551+2		INT 0543
STO •1		INT 0544
LDQ	CALCULATE NEW	INT 0545
FMP +1	VALUE OF P	INT 0546
FAD +1		INT 0547
STO •1		INT 0548
FDP DE2F+0552+2		INT 0549
STO COMMON+000		INT 0550
LDQ •1	CALCULATE NEW	INT 0551
FMP DE2F+0553+2	VALUE OF R	INT 0552
FAD COMMON+000		INT 0553
STO COMMON+000		INT 0554
TZE DE2F+0492		INT 0555
ARS 27		INT 0556
STO COMMON+001	TEST VALUES	INT 0557
CLA +1		INT 0558
TZE DE2F+0492	TO	INT 0559
SSP	DETERMINE	INT 0560
ARS 27	DETERMINE	INT 0561
SBM COMMON+001	IF SHIFTING	INT 0562
TM1 DE2F+0492	It Suit time	INT 0563
TZE DE2F+0492	IS NECESSARY	INT 0564
STA DE2F+0489		INT 0565
STA DE2F+0490 CLA COMMON+000		INT 0566
		INT 0567
ARS ALS		INT 0568 INT 0569
STO COMMON+000		• • • • • • • • • • • • • • • • • • • •
CLA +1	CALCULATE NEW	INT 0570
FAD COMMON+000	VALUE OF Z	INT 0571 INT 0572
STO 11		INT 0573
TRA DE2F+0558+2		INT 0574
LDQ +1		INT 0575
FMP DE2F+0556+2	CALCULATE	INT 0576
STO +1		INT 0577
LDQ +1	NEW VALUE	INT 0578
FMP DE2F+0555+2		INT 0579
FAD +1	A. A.	INT OSSO
STO +1	OF 9	INT 0581
LDQ COMMON+000		INT 0582
FMP DE2F+0554+2		INT 0583
FAD +1		

```
INT 0584
STO .1
TIX DE2F+0463.1.1
                           TEST N
                                                                    INT 0585
                                                                    INT 0586
CLA
                           INCREASE X
                                                                    INT 0587
FAD DE2F+0557+2
                                                                    INT 0588
STO
SXD DE2F+0557+2
                                                                    INT 0589
                       FIND DERIVITIVES
                                                                    INT 0590
TSX +4
                                                                    INT 0591
LXD DE2F+0557+2
                           TEST PASS NO.
                                                                    INT 0592
TIX DE2F+0462+2+8
                           RESTORE
                                                                    INT 0593
LXD DE2F+0553.1
                                                                    INT 0594
LXD DE2F+0554.2
                          INDEX
                                                                    INT 0595
LXD DE2F+0555.4
                          REGISTERS
                                                                    INT 0596
TRA 1,4
                                                                    INT 0597
DEC 0
DEC 2.
                                                                    INT 0598
DEC -1.
                                                                    INT 0599
                                                                    INT 0600
DEC 3.
                                                                    INT 0601
DEC -.5
                                                                    INT 0602
DEC 1.
                                                                    INT 0603
DEC
                                                                    INT 0604
TRA DE2F+0496
                                                                    INT 0605
DEC 0
DEC 2.
                                                                    INT 0606
                                                                    INT 0607
DEC -.5
                                                                    INT 0608
DEC -1.
                                                                    INT 0609
DEC -5
                                                                    INT 0610
DEC -3.
                                                                    INT 0611
DEC 0
                                                                    INT 0612
TRA DE2F+0558
                                                                    INT 0613
DEC -.5
                                                                    INT 0614
DEC 1.
                                                                    INT 0615
DEC 0
                                                                    INT 0616
DEC -1.
                                                                    INT 0617
DEC 0
                                                                    INT 0618
DEC 1.
                                                                    INT 0619
DEC
                                                                    INT 0620
TRA DE2F+0496
                                                                    INT 0621
DEC 2.
                                                                    INT 0622
DEC 6.
                                                                    INT 0623
DEC 1.
                                                                    INT 0624
DEC 3.
                                                                    INT 0625
DEC --5
                                                                    INT 0626
DEC -3.
                                                                    INT 0627
DEC 0
                                                                    INT 0628
TRA DE2F+0496
                                                                    INT 0629
PZE 32 ..
                                                                    INT 0630
PZE 1 ..
                                                                    INT 0631
BSS 5
                                                                    INT 0632
CLA .1
                                                                    INT 0633
FDP DE2F+0556+2
                                                                    INT 0634
STQ .1
                                                                    INT 0635
TRA DE2F+0499
                           SAVE INDEX REGISTERS
                                                                    INT 0636
SXD DE2F+0553.1
```

```
INT 0637
                               FROM MAIN PROGRAM
      SXD DE2F+0555.4
                                                                           INT 0638
      CLA 1,4
                                                                           INT 0639
      STA DE2F+0553
                                                                           INT 0640
      STA DE2F+0570
                                                                           INT 0641
      ARS 18
                                                                           INT 0642
                             SET ADDRESS OF
      STA DE2F+0512
                             DERIVITIVE ROUTINE
                                                                           INT 0643
      STA DE2F+0603
                                                                           INT 0644
                             STORE VALUE
      CLA
                                                                           INT 0645
                             OF N
      STO DE2F+0556
                                                                           INT 0646
      CLA DE2F+0553
                                                                           INT 0647
      ADD DE2F+0552
                                                                           INT 0648
                             STORE ADDRESS
      STA DE2F+0508
                                                                           INT 0649
                             OF X
      STA DE2F+0510
                                                                           INT 0650
      ADD DE2F+0552
                                                                           INT 0651
                             STORE ADDRESS
      STA DE2F+0466
                                                                           INT 0652
                             OF H
      STA DE2F+0457
                                                                           INT 0653
      ADD DE2F+0552
                                                                           INT 0654
      ADD DE2F+0556
                                                                           INT 0655
      STA DE2F+0479
                              STORE ADDRESS
                                                                           INT 0656
                             OF Y
      STA DE2F+0492
                                                                           INT 0657
      STA DE2F+0494
                                                                           INT 0658
                              STORE ADDRESS
      ADD DE2F+0556
                                                                           INT 0659
      STA DE2F+0467
                              OF DERIVITIVE
                                                                           INT 0660
      ADD DE2F+0556
                                                                           INT 0661
      STA DE2F+0605
                                                                           INT 0662
                              STORE
      STA DE2F+0560
                                                                           INT 0663
      STA DE2F+0558
                                                                           INT 0664
                              ADDRESS
      STA DE2F+0472
                                                                           INT 0665
      STA DE2F+0496
                                                                           INT 0666
                              OF
      STA DE2F+0498
                                                                           INT 0667
      STA DE2F+0501
                                                                           INT 0668
                              0
      STA DE2F+0502
                                                                            INT 0669
       STA DE2F+0505
                                                                            1NT 0670
       STA DE2F+0506
                                                                            INT .0671
       ADD DE2F+0556
                                                                            INT 0672
       STA DE2F+0463
                                                                            INT 0673
                              STORE ADDRESS
       STA DE2F+0465
                                                                            INT 0674
       STA DE2F+0468
                                                                            INT 0675
                              OF P
       STA DE2F+0469
                                                                            INT 0676
       STA DE2F+0499
                              FIND INITIAL DERIVITIVES
                                                                            1NT 0677
       TSX +4
                                                                            INT 0678
       LXA DE2F+0556+1
                                                                            INT 0679
                              SET ORGINAL O
       STZ +1
                                                                            INT 0680
                                  TO ZERO
       TIX DE2F+0605+1+1
                                                                            1NT 0681
                                   RESTORE INDEX
       LXD ~E2F+0553+1
                                                                            INT 0682
                                 REGISTERS
       LXD DE2F+0555+4
                                                                            INT 0683
       TRA 2.4
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       END
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H. SUBROUTINE RINDEX

This subroutine permits the calculation of the refractive index, its spatial derivatives, and the absorption coefficient, all as functions of the local atmosphere and its state of ionization. This subroutine also permits the output of "debugging data" which is called the R vector whose components are defined in Table 3. See subroutine OUTPUT for additional information.

$$R(2) = PXR = \frac{\partial X}{\partial r}$$

$$R(3) = PXTHET = \frac{\partial X}{\partial \theta}$$

$$R(4) = PXPHI = \frac{\partial X}{\partial \phi}$$

$$R(5) = Y$$

$$R(6) = YR = Y_r$$

$$R(7) = YTHETA = Y_{\theta}$$

$$R(8) = YPHI = Y_{\phi}$$

$$R(9) = PYR = \frac{\partial Y}{\partial r}$$

$$R(10) = PYTHET = \frac{\partial Y}{\partial \theta}$$

$$R(11) = PYPHI = \frac{\partial Y}{\partial \phi}$$

$$R(12) = Z$$

$$R(13) = PZR = \frac{\partial Z}{\partial r}$$

$$R(14) = PZTHET = \frac{\partial Z}{\partial \theta}$$

$$R(15) = PZPHI = \frac{\partial Z}{\partial \phi}$$

$$R(16) = COSPSI = \cos \psi$$

$$R(17) = SINPSI = \sin \psi$$

$$R(18) = YSI = Y \sin \psi$$

$$R(19) = YCI = Y \cos \psi$$

R(1) = X

Table 3. Nomenclature Describing the R Vector.
(Page 1 of 4)

$$R(21) = TE2 = (YSI)^{4} + 4TE1(YCI)^{2}$$
 $R(22) = TE3 = 8(YCI)^{2} Z(1 - X)$
 $R(23) = R2S = R_{S}^{2}$
 $R(24) = R1S = R_{S}$
 $R(25) = THET2S = 2\theta_{S}$
 $R(26) = THET1s = \theta_{S}$
 $R(27) = S1 = S_{1}$
 $R(28) = S2 = S_{2}$
 $R(29) = D1 = d_{1}$
 $R(30) = D2 = d_{2}$
 $R(31) = TE4 = d_{1}^{2} + d_{2}^{2}$
 $R(32) = TE5 = 2X[Zd_{1} + (1 - X)d_{2}]/TE4$
 $R(33) = TE6 = 1 - [2X(1 - X)d_{1} - Zd_{2}]/TE4$
 $R(34) = R2M = R_{M}^{2}$
 $R(35) = R1M = R_{M}$
 $R(36) = THET2M = 2\theta_{M}$
 $R(37) = THET1M = \theta_{M}$
 $R(38) = AM1 = M_{1}$
 $R(39) = AM2 = M_{2}$
 $R(40) = TE7 = M_{1}d_{1} - M_{2}d_{2}$

Table 3. Nomenclature Describing the R Vector. (Page 2 of 4)

R(41) = TE8 =
$$M_1d_2 + M_2d_1$$

R(42) = AO = a_0
R(43) = BO = b_0
R(44) = TE9 = $S_1^2 + S_2^2$
R(45) = A4 = a_4
R(46) = B4 = b_4
R(47) = A5 = a_5
R(48) = B5 = b_5
R(49) = PNPX = $\partial \mu / \partial X$
R(50) = TE10 = $(\sin \psi)^2 (YSI)^2 + 2TE1 \cos^2 \psi$
R(51) = TE11 = $4Z(1 - X)\cos^2 \psi$
R(52) = A6 = a_6
R(53) = B6 = b_6
R(54) = PNPY = $\partial \mu / \partial Y$
R(55) = A7 = a_7
R(56) = B7 = b_7
R(57) = PNPZ = $\partial \mu / \partial Z$
R(58) = A8 = a_8
R(59) = B8 = b_8
R(60) = TE13 = $1/(2\cos^2 \theta + \frac{1}{2}\sin^2 \theta)$

Table 3. Nomenclature Describing the R Vector. (Page 3 of 4)

R(61) = A1 = a₁
R(62) = B1 = b₁
R(63) = A2 = a₂
R(64) = B2 = b₂
R(65) = TE12 =
$$(W_1)^2$$
 YSI
R(66) = PPSIPT = $\partial \psi / \partial \theta$
R(67) = PPSIPR = $\partial \psi / \partial \tau$
R(68) = PPSIPP = $\partial \psi / \partial \phi$
R(69) = TE14 = $\sqrt{\sigma_r^2 + \sigma_\theta^2 + \sigma_\phi^2}$

Table 3. Nomenclature Describing the R Vector. (Page 4 of 4)

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I. SUBROUTINE ELECTX

The whole process of determining the free normalized electron density and its spatial derivatives at any point in space is explored in this subroutine. For the case under consideration this is a simple sphere where the values in Equation 74 are represented by the following components of the W vector

$$W(28) = \mathcal{L}; W(34) = A; W(35) = n$$

J. (r) (D)	RM 611	MP-32	N							
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ie+*N DEC. 22, 1960 From Radius = W(36)					T+28RP>					
TX GIVEN BY NESP/RUNGESSN HAVING NE = CONSTRNT FROM F (X, PXR, PXTHET, PXPH I) (2), UCITY, #C250)	/#(3)>+3,184983E9 ,11 (5)>				*XBRP+5T*5P*YBRP+CT*ZBRP>	12.24.1464	W(3)+5@RTF(X)	0.0.0.0.0.0.0		
E SPHERE ELECTX INNER SPHERE UTINE ELECTX SION RECORD. V. W.	TER! = (CBC34) /BC3) /BC3) /BC3 /BC3 /BC36) /BC36 /BC36) /	FYPHI = 0.0 GO TO 40 ST = SIMF (V(S))	# COSF (V(5)) # SIMF (V(6)) # COSF (V(6)) # V(4) = 1			in bill	FXM = IRSEPANTM FXTMET = TR3-PFRTT PXPMI = TR3-PFRTP W(36) = CX-W(3)/3, 184983E9	RETURN DDC0, 1,0, 1,0,0,0,0,0,0,0,0		
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J. SUBROUTINE BIGR

This subroutine is used to calculate the distance from the center of the ionizing source to the spatial point r, θ , Ψ at which the electron density and spatial gradients are desired. For the simple sphere this distance is $W(28) = \mathcal{H}$. In addition to this it calculates the angle that \mathcal{H} makes with respect to the vertical distance passing through the center of the ionizing source.

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K. SUBROUTINE MAGY

The normalized earth's magnetic field and its spatial derivatives are calculated in this subroutine. In this program it is represented by an assumed magnetic dipole field with the dipole located at the center of the earth.

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L. SUBROUTINE COLFRZ

The normalised collision frequency and its derivatives at a spatial point are determined in this subroutine. In this presentation it is assumed that collision frequency varies exponentially with height. One should refer to the definition of the W vector where the required coefficients for the height stratifications are defined.

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M. SUBROUTINE RCOORD

This subroutine permits the transformation from the earth centered geomagnetic spherical coordinate system, to the radar coordinate system.

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N. SUBROUTINE OUTONE

In this presentation this subroutine is used to list the input data and the initial conditions that define the ray tracing problem on Tape Unit 6. Table 5-1 illustrates this output by the statement of the input RECORD and the next ninety-one words of data which define in order the first 70 components of the W vector, followed by the first 21 components of the V vector. This subroutine can be greatly improved.

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W 32438 77266	EFN IFN LOC 9 6 00000 6 16 00035		DEC OCT SOURCE PROGRAM	22 DEC OCT 23 58 00072 H LIBRARY.		
77461 U 32549 77445 W 32438 77266	EFN IFN LOC 7 5 60000 5 15 00034 17 USED BY PROGRAM	IES IN TRANSFER VECTOR	OCT DEC OCT DEC OCT DEC TO SOURCE PROGRAM	00111 824 BG 00120 22 NTS TO SUBROUTINES NOT OUTPUT FROM LIBRARY.		
RECORD 32561 77461	EFN 1FN LOC 4 4 00000 3 14 00033 8 29 00065 STORAGE NOT	0EC 0CT 32187 76673 LOCATIONS OF NAMES	DEC OCT CSTRS 0 00000	8)7 73 00111 ENTRY POINTS TO SE		
	EFN 1FN LOC 3 3 9 00021 8 22 00050	0EC 0CT 81 00121	CFIES DEC 0CT	8>9 67 00103 67 00103 787A5 (F1L)		

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O. SUBROUTINE OUTPUT

This subroutine will presently yield the output data on Tape Unit 6 that is summarized in Tables 5 and 6. Following the information that is the result of subroutine OUTONE, Tables 5-1 and 5-2 illustrate the information obtainable after each integration. Beginning with the first word this information has the following meaning:

Integration number W(70)
Vector components V(2), V(3)
Height above earth surface (km)
Angle \theta in degrees
Angle \theta in degrees
Vector components V(7) through V(21)
Vector components W(1) through W(11)
Vector components W(28), W(29), W(38), W(39)
Vector components W(55) through W(70)
Vector components W(80) through W(90)

Under certain conditions of vector components W(68) and W(69) the R vector described in subroutine RINDEX will be listed between each of these sets of data for each time that the RINDEX subroutine is entered.

In addition to this data, the data summarized in Table 6 is listed on Tape Unit 10.

oʻ	SUBROUTINE OUTPUT JANUARY 21, 1961	18M-7090	Andre de la cineta del cineta de la cineta del cineta de la cineta del cineta de la		
-					(.1
•	COMMON RECORD, V. B. N. XN. VN. ZN. G DIMENSION RECORD(12), V(1)11, V(250)				
	COLL PCORPD) 1 1 1 1 1 1 1 1 1 1 1 1 1	(F)
•	IFCUCTOS 3.3.4			į	7
•	W(70) = W(70) + 1.0				(U
	(0) no (0) n = 1	700			
m	TE1 = V(5)+57,29578	(Z++ (Z)A - ;			
•	TE2 = V(6)+57.29578				Ï
	163 H V(4)-8(19) M(56)-(2,9979)65+U(11))-8(55)				•
					:7
1					
• •• •	BRITE OUTPUT TAPE 6,8,8(70),V(2),V(3),TE	,, V(2),, V(3), TE3, TE1, TE2, V(7), V(8)			1
<u> </u>	•				
9		•			ı
•		(), I=9,21), CWCI), I=1,11), WC28), WC29),			t
	18(55), 8(57), (8(1), 1809, 70), (8(1), 1860, 90) 1F(11) 10, 10, 30	2			
<u>.</u>	1PE 10.11, CR	S			
:	WRITE OUTPUT TAPE 6,11, (RECORD(I), I=1,12)	23		•	;
•	WRITE OUTPUT TAPE 10.12				
12	FORMATCHO IN.SX. SHSLANTR, 6X. 4HV(2), 6	R, 6X,4HV(2),6X,13HV(11)+C-4(60),4X, F,9X,6M3; T-E,7X,7M9M6; F,0,7X,6M6; T-0)		•	
30					
31	UTPUT TRPE 10.32. IN.	#C60>, V(2), #(83), #(82), #(62), #(81), #(61	,		
32	FORMOTCH . 14.1P0E14.70				
	IF(II-50>40, 33, 33		• • • • • • • • • • • • • • • • • • •		!
# ?					;
	END(0, 1.0, 1.0, 0.0, 0.0, 0.0, 0.0, 0.0)				
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	2 0CT		2000	18 00052	42 00167 57 00253	1			C 0CT	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	EC 0CT 229 00345	1	C 0CT			
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ENTS	0CT 77445	OCTAL LOCATIONS			00163				100	EQUIVALENCE OR COMMON SENTENCES	0CT 00346	, E	001 00335 0 00276			
COMMON STATEMENT	0EC v 32549	문	EFN IFN	- 5	9 31 34)))	VALENCE OR	DEC TE2 230	SOURCE, PROGRAM	8)8 221 6) 190	IBRORY		
APPERRING IN CO	77461 76655	INTERNAL FORMULA NUMBERS	707	00 000 00 020	00124 00220	ROM		VECTOR	00001	IN DIMENSION. EQUI	0CT 00347	MOT GPPEARING IN SO	0CT 00332 00272	not. Qutput .erom l. Jbrorx		
FOR VARIABLES APP	DEC RECORD 32561 ZN 32173		F		3. 28	USED BY PROGRAM		IN TRANSFER	OEC CSTH)	APPEARING IN DI	DEC 133	SYMBOLS NOT RP!	B)C 218	SUBROUTINES NOT		
LOCATIONS FOR VA	DCT 76674 RE 76664	WITH CORRESPONDING	201	00000 00016	00077	STORAGE NOT US		NAMES			00150	OCATIONS FOR SYM	00305 00305 00261	.POJNTSTQ.SUBROI		
STORAGE LOCK	DEC N 32168 VN 32160	ULA HUMBERS U	IFR	10 M	30 24	S	DEC 32156		DEC	OR VARING	DEC 11 232	-4	DEC 197	EBTRY POJ		
ST		FORMULA	EFX		i			1	GIL	CATIONS F		STORAGE	6 510	Ğ.		
	EC 0CT 165 76645 187 76673	EXTERMAL	15N	- 4	23 00074 46 00174		DEC OCT 234 D0352		DEC OCT 0 00000	STORAGE LOCATIONS FOR VARIABLES NOT	DEC 0CT 233 00351		DEC OCT 222 00336 226 00342	CSTIO		
	DEC G 32165 km 32187	;		; - c	, r o	?	ō``	:	RCOORD D	i	5		0 1100	RCOORD		

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P. INPUT-OUTPUT

Table 4 illustrates the format that is necessary for the input data. Tables 5 and 6 illustrate the format of the output data that is currently obtained by use of Subroutines OUTONE and OUTPUT. These two subroutines can be easily modified. As they are presented in this report their primary purpose was to collect "debugging" data.

Table 4. Input Data for a Spherical lonospher

ì	•	•	 Ĭ	•	• ;	5	= 1	S	 {	 -	<u> </u>	RM 61TMP-32	
.0	10 7.2082097E-01	22 1.0000000E 00	02-4.6394721E 03	1.200000E 03	1 6.	0.	00 0.	2.28020518 00	IS 2, 2802051E 00	0.	.3278773E-01-8.98 590 70E-01	2 1,2639290E-04 6,380000E-09 1,3316441E-20 11 5,5473943E-03 18 5,4883252E-11 15-2,2964180E-03	
· o	.6059238E 00 1.5707963E 00	4.6059238E 00 3.0000000E	6.6569111E 03-4.9581701E 0	0.	5.00000000	55717E 03 0.	3.000000E	0.	9865271E-01-2.3434196E-08	.0	2.6389999E G2 4.3878773E-0	4.387379E-01-1.4132452E-04-4.9081066E-12 1.2639230E-09-09-09-09-09-09-09-09-09-09-09-09-09-	•
0.	5.	7.9412480E-31	6.65	0.	.0.0	02-4.7486789E 33 4.195571	·0.	0 9.9999€9E-05 0.	4.3866078E-01-3.	0.	4.8673434E ():	3. 2802163E 00- 2. 00000000E 00- 3. 2802163E 00- 3. 0000000E	
000 6 09 0.	0	112E 03 1.00053333E 00	Ö	0.00 0.00	330E 01 1.0003000E 00	-5.074=736E 0	0	code 00-1.0000000E 00	302E-01 4.6059233E 00	0.	000E_00_1.43835+5E_00	917E-05 0, 3333333E-06 0000E 00 3, 3333333E-06 00 3, 3333333E-06 00 0000 0000 000 000 000 000 000 000	
5.280000	9	6.3360	. 0.	. 0000000E 33] 2000000	≶.999998E-06 5.000000	j.). 	3579108E G3 8.496530		0. 2.2802051E 00 1.0.02020	08 3,2802031E CO 1,0839917 05-1,2951176E-12 1,0000000 110-3,4459441E-19 8,272174 02-3,459443E-01 5,7859963 03-1,0000000E OC 9,999999 03-1,0000000E OC 9,999999	
1.55000000E 00 0	· · · ·	+.5378560E- 1 0	5.4677431E U1 3	+.7480446E 53 1	9	1.6665300 E 30 3	ō	.0.	1.0000000E 00 C	7.60068 37£- 0310	0. 1. CCD0200E 20 3	-2.3425490E-08-3.2802051E 6.2011474E-05-1.2893176E- 5.0457822E-19-2.4638431E- 5.601182E 02-3.6324448E- 1.0000000E 00-1.0000000E 0.0.00000E	

Output Data for "Debugging" Purposes from Tage 6.

RM 61TMP-32		
SYMPLE SPHERE 10#433/R4412 Z = 0.0 V VERV SHALL FEB. 3:1961 E=2603	0934917E-05 0. 4.3916866E-01- 0934917E-05 0. 333333E-06 0. 7141767E-28-4.8977747E-13 5.318962E-09- 9745170E 04 1.1203736E 00 6.2802376E 00- 9745170E 04 1.1203736E 00 6.2802376E 00- 9795959E-05 4.000000E 00 0. 0000000E 00 3.1945190E 00 4.8641053E 01- 0000000E 00 3.133333E-06 0. 6.392959E-05 0. 11452417E 00 7.280249E 00- 0000000E 00 3.333333E-06 0. 0000000E 00 3.6333333E-06 0. 0000000E 00 3.6333333E-06 0. 0000000E 00 3.6333333E-06 0. 0000000E 00 3.6333333E-06 0. 0000000E 00 3.6333333E-06 0. 0000000E 00 3.6333333E-06 0. 0000000E 00 3.6333333E-06 0. 0000000E 00 3.6333333E-06 0.	8.2802662E 00 0. 2.5957436E 01 3.00000CUE UIJ U. 2.5632419E-03-6.1035156E-03-5.005-5
		0

Table 5-2. Output Data for "Debugging" Purposes from Tape 6.

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9C105-#C605	87	. 2823296E-05	•	84531E	. 3728628-UJ	n w	77440E-05		801770E-03	19298E-06	64551E-05	862228E-05	54880E-05	96649E-05	ĸ	78601E-05	60553E-05	89948E-05	. 1484833E-05	02E-06	8881846-03	7 41549685	444092E-05	757202E-05	1.1444092E-05	.29425U3E-U3	1923340E-05	6239624E-05	2.6702881E-05	4.19616706-05	- ·	4.9591064E-05	•		ď	3.8146973E-06 2.	"[:	ŗα	17578E-0	.8146973E-0	.6702881E-0	.6702881E-	32275E-	1.9073486E-05	604001E
(11) *C-1(60)	2964180E-03-1	ŏ	12570	4077635E-03-	2263	6919		0.00E-0	0.3287E-0.	ζ:	0-300E-0			7668E-0	24686E-0	95245E	97141E-0	36940E	37487E-02-	75951E-02-	20613E	10047	• (N	50848E-02-	.7742081E-02-	94012E-01-	04662E-01-	49960E-01-	78316E-01-	08960E-01-	340736-0		6113	61130E	04398E-	28873E-	14000	- 10 27 27	06.30SE	03036E	22006E-	.7322006E-01	.6126362E-01	.4251938E-01-	. 1312332E-01
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UC25	3.2802051E	1.2802051E	3.2802051E	5. 2802051E	7.2802051E	3.2802051E	1.0280205E	1. 2280205E	1.4280205E	1.6280205E	1.8280203E	4.050000E	2.8280203E	3,2280205E	3.6280205E	4.0280204E	4.4280205E	5.2280204E	6.0280204E	6.8230205E	7.6280205E	8.4280203E	1.0828020E	1.2428020E	1.4028020E	1.5628020E	1.7228020E	2.2028020E	2.5228020E	2.8428020E	3.16280206	3.4828020E	3. 5028020C	4.1228020	4.28280206	4.4428020	4.6028020	4.1228020	4.2828020	4.3628020	4.4428020	4.44280206	4.4828020	4.5228020	4.5628020
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Table 6-1. Output Data from Tape 10.

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